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Structural Report

F54S

17341

for the system by

Global Truss

Furong Industrial Area
Shajing Town

Baoan District Shenzhen China

compiled by:

A. Rempel

Aachen, 30.06.2017



This Structural Report includes pages

1 - 32

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ANNEXES

Annexe A: Comparative Calculation influence of horizontal levels

S. A1 – 7

Annexe B: Drawings F54S

F54S050-Model, F54S100-Model, F54S150-Model, F54S200-Model,
F54S250-Model, F54S300-Model, F54S350-Model, F54S400-Model,
F54S450-Model, F54S500-Model



1 PRELIMINARY NOTES

1.1 Basics

The currently applicable regulations and standards, in particular:

DIN EN 1991-1	Actions on structures (Eurocode 1)
DIN EN 13814	Fairground and amusement park machinery and structures
DIN EN 13782	Temporary Structures – Tents
DIN EN 1993-1	Design of steel structures (Eurocode 3)
DIN EN 1999-1	Design of aluminium structures (Eurocode 1)

1.2 Materials

Tubes	Aluminium EN AW-6082 T6
Bolts	42CrMo4 8.8

1.3 General Remarks

This structural report is an structural calculation concerning a truss system produced by the company GLOBAL TRUSS. The truss type goes by the name F54S.

The structural report is the basis for the certification by TÜV based on EN 1999-1.

The truss system is part of a "modular construction system" with the different truss lengths

500mm, 1000mm, 1500mm, 2000mm, 2500mm, 3000mm, 3500mm, 4000mm, 4500mm and 5000mm.

The Trusses consist of two upper and two lower main chords (round tube 50 x 4mm), which are arranged in a quadratic shape.

The trusses also consist of welded diagonal bracings (round tube 30 x 3mm) at the sides and welded vertical struts (round tube 50 x 4mm) between the nodes of the diagonal bracings in the upper and lower level.

The distance between the system lines of the mainchords is 47 cm in vertical- and 47 cm in horizontal direction.

The trusses are connected with couplers consisting of female fittings, connectors and bolts.

The allowable loads are listed in tables (see chapter 6).

The verification of the single parts is done according the safety concept of EN 1990 with a partial safety factor of the loading side of 1.50 for payloads.



For applications which can be calculated on the basis of other codes, the partial safety factors can be adjusted (for example temporary structures acc. EN 13814, $\gamma_F = 1.35$ for payloads).

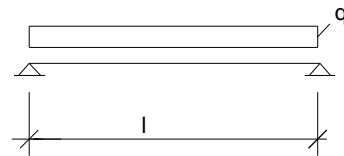
To use the resulting allowable loads with British Standard (BS) and ANSI, the allowable loads listed in tables have to be multiplied by 0.85.



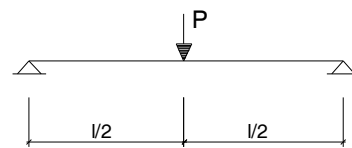
1.4 Geometry and Loadings

The following loadcases are taken into account

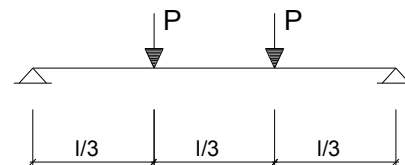
uniformly distributed load (UDL)



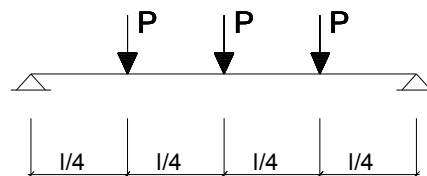
Single-load in 1/2 point



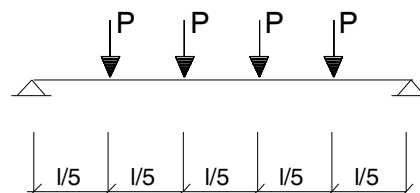
Single-load in 1/3 point



Single-load in 1/4 point



Single-load in 1/5 point

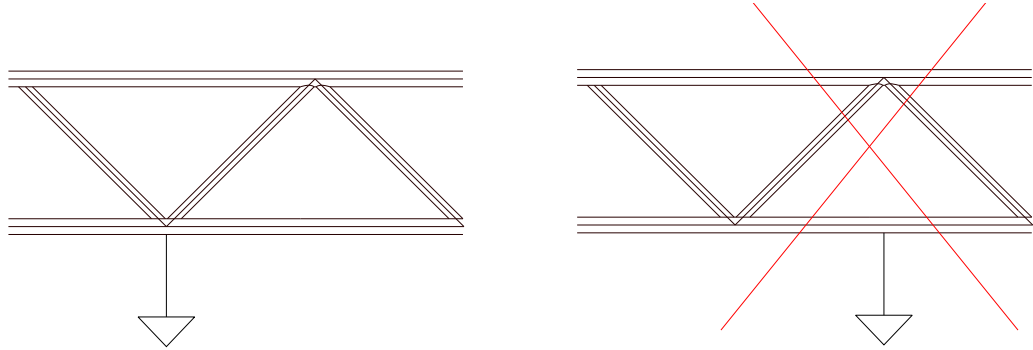


The selfweight of the truss is approx. 16 kg/m

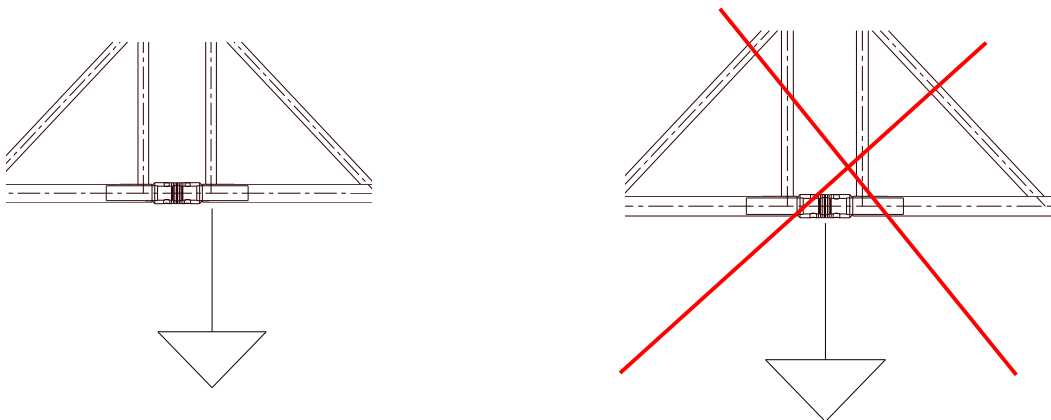


For the application of the calculated allowable loadings the following rules have to be regarded:

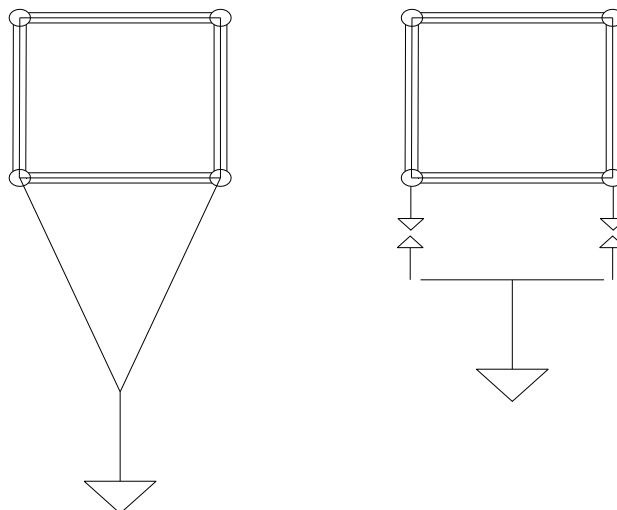
Large loads have to be applied at the nodes or have to be distributed by appropriate constructions.



Loads at the middle of the couplers are not allowed.

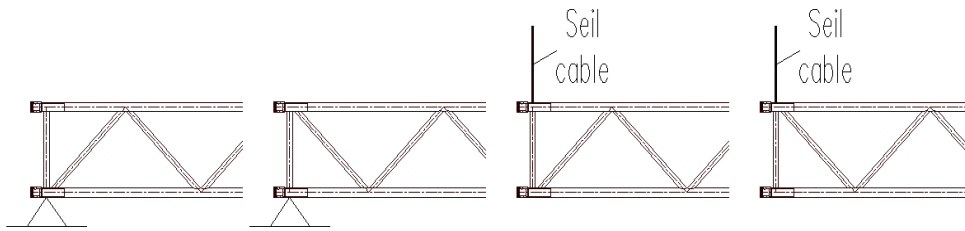


All loads have to be distributed equally to both chords.

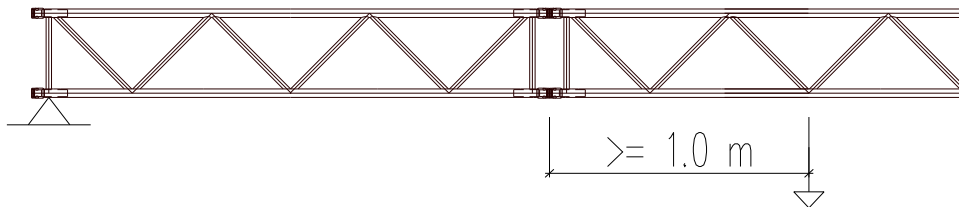




For the support or suspension there are the following possibilities:



When restricting the position of the coupler (see chapter 4, case 2), the coupler has to be located $> 1,0$ m from the loading point into the direction of the support.





2 SYSTEM

Drawings F54S.

F54S050-Model, F54S100-Model, F54S150-Model, F54S200-Model,
F54S250-Model, F54S300-Model, F54S350-Model, F54S400-Model,
F54S450-Model, F54S500-Model

see annex



3 SECTION- AND MATERIAL PROPERTIES

Properties Tubes						
	D	t	A	I	Wel	i
	[mm]	[mm]	[cm ²]	[cm ⁴]	[cm ³]	[cm]
main chords	50,0	4	5,78	15,41	6,16	1,63
vertical Bracing	30	3	2,54	2,35	1,56	0,96
horizontal Struts	50	4	5,78	15,41	6,16	1,63

Truss geometry				
distance axes main chords	vertical	ev	47	[cm]
	horizontal	eh	47	[cm]
min. gradient bracing	vertical	α	44,6	[°]
	horizontal	α	-	[°]
Properties truss-Section				
A	=	4 x A _G	=	23,12 [cm ²]
I _{yy}	=	4 x I _G + 4 x A _G x (ev/2) ²	=	12830,81 [cm ⁴]
I _{zz}	=	4 x I _G + 4 x A _G x (eh/2) ²	=	12830,81 [cm ⁴]
I _t	=		=	876,00 [cm ⁴]
i _y	=	(I _{yy} /A) ^{1/2}	=	23,56 [cm]
i _z	=	(I _{zz} /A) ^{1/2}	=	23,56 [cm]
Index G : section properties main chord				

Calculation of torsion moment of inertia:

A torsion moment of 1,0 kNm is applied. The torsion moment of inertia is calculated with the resulting deformations.

$$\text{Average deformation } u = (6,3 + 6,2 + 4,5 + 4,4) / 4 = 5,35 \text{ mm}$$

$$\Rightarrow \vartheta = 5,35 / (470 \cdot \sqrt{2} / 2) = 0,0161$$

$$M_T = 1,0 \text{ kNm} = 100 \text{ kNcm}$$

$$G = 27000 \text{ N/mm}^2 = 2700 \text{ kN/cm}^2$$

$$l = 3,81 \text{ m} = 381 \text{ cm}$$

$$I_T = M_T \cdot l / (G \cdot \vartheta) = 100 \cdot 381 / (2700 \cdot 0,0161) = 876,6 \text{ cm}^4$$

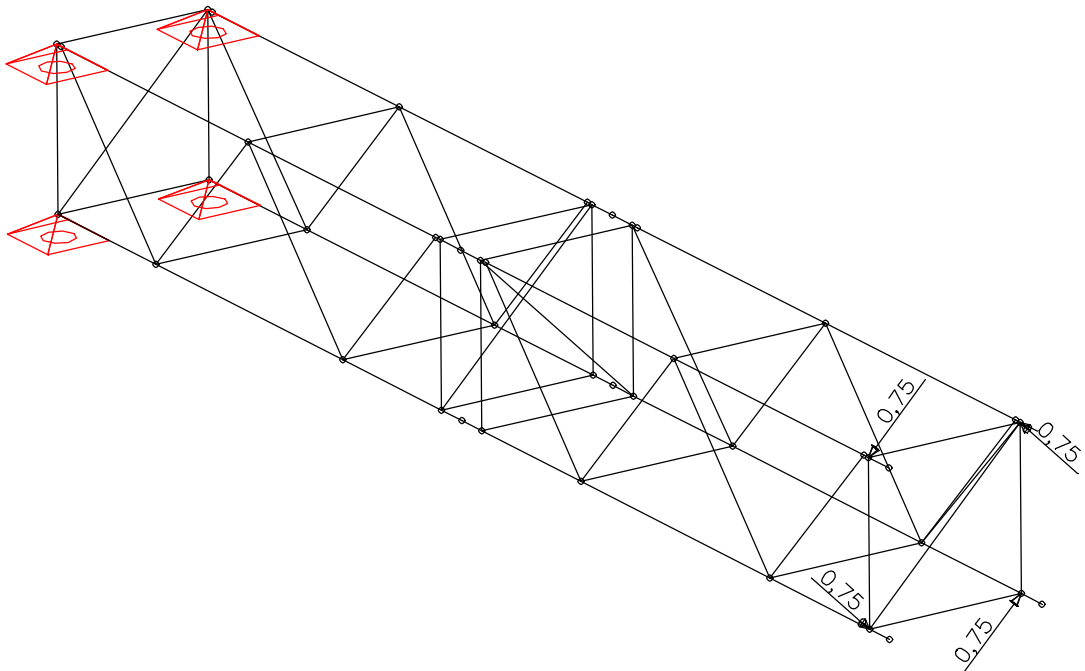
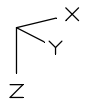


17341 - F54S

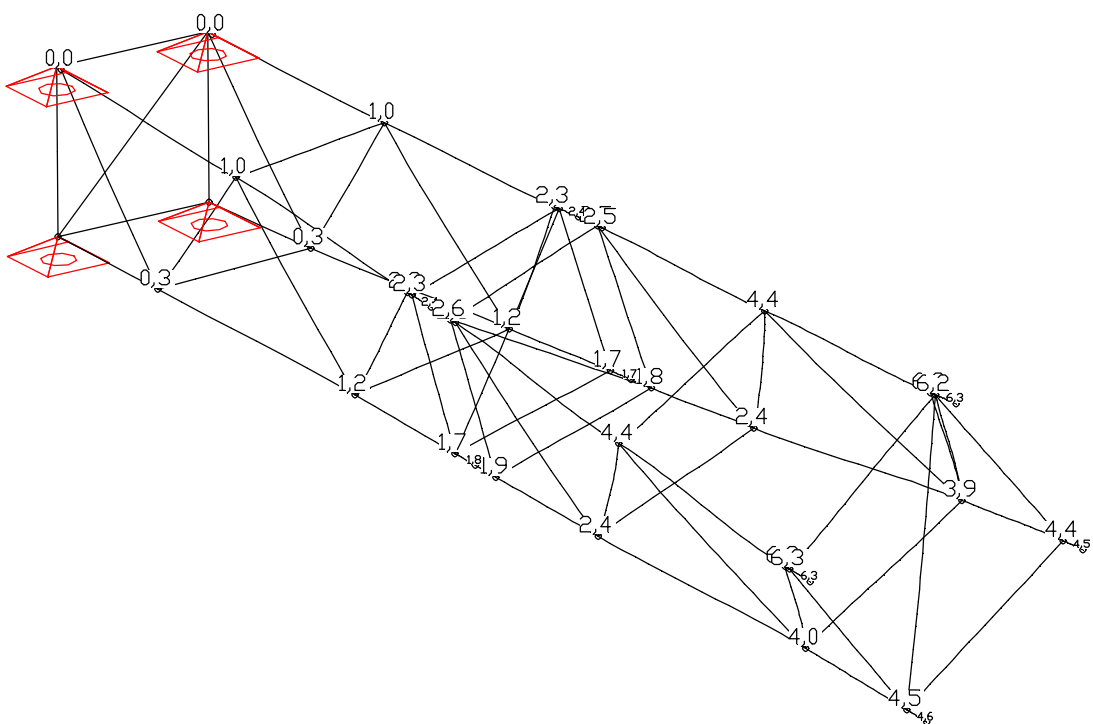
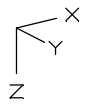
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Torsion Moment of Inertia

M 1 :



LC 1: Load



Deformations u; LF 1



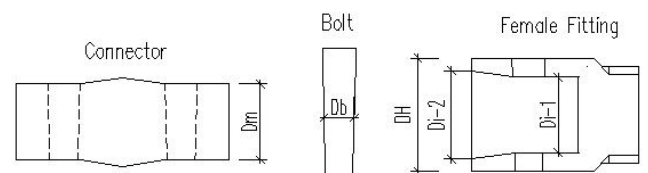
Material properties

Chords and bracing	EN AW 6082 T6 (AlMgSi1)		
allowable stress acc. to EN-1999-1-1			
Partial safety factors material			
YM1=	1,10	Buckling class / BC=	A
YM2=	1,25		
0,2%-Proof Strength	ultimate tensile strength		
fo t≤5mm=	250 [N/mm²]	fu t≤5mm=	290 [N/mm²]
fo t>5mm=	260 [N/mm²]	fu t>5mm=	310 [N/mm²]
fo,haz=	125 [N/mm²]	fu,haz=	185 [N/mm²]
Strength of welding seams	fw=		190 [N/mm²]
Factor for HAZ-values for TIG-welding:	0,8		

Bolt	42 CrMo (8.8)
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Connector	EN AW 2011 T6 (AlCuBiPb)		
0,2%-Proof Strength	ultimate tensile strength		
fo>	230 [N/mm²]	fu>	310 [N/mm²]

Female fitting	EN AW 6082 T6		
allowable stress acc. to EN-1999-1-1			
partial safety factors material			
YM1=	1,10		
YM2=	1,25		
0,2%-Proof Strength	ultimate tensile strength		
fo=	250 [N/mm²]	fu=	290 [N/mm²]





Section- and material properties of the chord tubes

Material	E=	70000	[N/mm ²]	
	f _o =	250,00	[N/mm ²]	
	f _o /Y _{M1} =	227,27	[N/mm ²]	
	f _{o,haz} =	125,00	[N/mm ²]	
	f _u =	290,00	[N/mm ²]	
	f _u /Y _{M2} =	232,00	[N/mm ²]	
	f _{u,haz} =	185,00	[N/mm ²]	
cross section	f _{u,haz} /Y _{M2} =	148,00	[N/mm ²]	
	D ₀ =	50,00	[mm]	
	A=	5,78	[cm ²]	
	I=	15,41	[cm ⁴]	
Determination of section-clas	i=	1,63	[cm]	
	β=	10,61	[-]	3 · (D ₀ / t) ^{0,5} nach 6.10
	ε=	1,00	[-]	(250 / f _o) ^{0,5}
	Section class=	2		acc. chapter 6.1.4.4
Coefficients for buckling	BC=	A	[-]	
	α=	0,20	[-]	
	λ ₀ =	0,10	[-]	
teff in heat affected zone	red-Faktor=	0,8	[-]	(WIG TIG)
	node with 2 bracing			
	D ₁ =	30,00	[mm]	
	D ₂ =	50,00	[mm]	
	U _{WEZ} =	153,54	[mm]	π / 2 · D ₀ + D ₁ / 2 + 2 · 30
	U _{Total} =	157,08	[mm]	
	t _{eff,o} / t=	0,41	[-]	[1 - (1 - red-Faktor · f _{o,haz} / f _o) · U _{WEZ} / U _{Total}]
	t _{eff,u} / t=	0,52	[-]	[1 - (1 - red-Faktor · f _{u,haz} / f _u) · U _{WEZ} / U _{Total}]

Section- and material properties of the bracing

Material	E=	70000	[N/mm ²]	
	f _o =	250,00	[N/mm ²]	
	f _o /Y _{M1} =	227,27	[N/mm ²]	
	BC=	A	[-]	
	α=	0,20	[-]	
	λ ₀ =	0,1	[-]	
cross section	D ₀ =	30	[mm]	
	A=	2,54	[cm ²]	
	I=	2,35	[cm ⁴]	
	i=	0,96	[cm]	



4 ALLOWABLE LOADING SINGLE COMPONENTS

Main chord in heat affected zone at coupler			
$NRd = A \times 0,8^* \times f_{u,haz} / Y_{M2} =$	68,44	[kN]	*(WIG π_G) local welding seam acc. chapter 6.2.9.3 (1)
Main chord in heat affected zone node with 2 bracing			
$NRd = A_{eff} \times f_o / Y_{M1} =$ (mit $A_{eff} = t_{eff,o} / t \times A$)	54,33	[kN]	local welding seam acc. Chapter 6.2.9.3 (2)
Buckling main chord bewteer nodes without bracing in the middle			
$sk =$	94,00	[cm]	
$N_{cr} =$	120,45	[kN]	
$\lambda^* =$	1,10	[-]	
$\phi =$	1,20	[-]	
$X =$	0,59	[-]	
$NRd = X \times A \times f_o / Y_{M1} =$	77,82	[kN]	acc. equation 6.49
Welding seam between chord and female conical coupler			
$f_w =$	190,00	[N/mm ²]	
$Y_{mw} =$	1,25	[-]	
$NRd = A \times f_w / Y_{M1} =$	87,86	[kN]	acc. equation 8.29

relevant for main chord tubes: **$NRd_G = 54,33$ kN**



Section- and material properties of the bracing			
Material	E=	70000	[N/mm ²]
	f _o =	250,00	[N/mm ²]
	f _o /Y _{M1} =	227,27	[N/mm ²]
	BC=	A	[-]
	α=	0,20	[-]
	λ ₀ =	0,1	[-]
cross section	D ₀ =	30	[mm]
	A=	2,54	[cm ²]
	I=	2,35	[cm ⁴]
	i=	0,96	[cm]
Bracing in heat affected zone			
$NR_d = A \times 0,8 \times f_{u,haz} / Y_{M2} =$		30,13	[kN] *(WIG πG) local welding seam acc. chapter 6.2.9.3 (1)
Buckling bracing	sk=	50,00	[cm] (buckling length = 0,75 x l)
	N _{cr} =	64,87	[kN]
	λ* =	0,99	[-]
	φ=	1,08	[-]
	X=	0,66	[-]
	$NR_d = X \times A \times G \times f_o / Y_{M1} =$		38,33
Welding seam between chord and female conical coupler			
f _w =		190,00	[N/mm ²]
Y _{mw} =		1,25	[-]
$NR_d = A \times f_w / Y_{M1} =$		38,68	[kN] acc. equation 8.29

relevant for bracing tubes: **NR_d = 30,13 kN**



Allowable normal force at coupler:

Bolt		
Material 42CrMo (10.9)	fy,bk=	64,00 [kN/cm ²]
	fu,bk=	80,00 [kN/cm ²]
Geometry	Db=	1,53 [cm]
	Ab=	1,84 [cm ²]
Allowable loading due to shearing acc. to EN 1999-1-1		
	$NRd = 2 \times 0,60 \times Ab \times f_{ub,k} / 1,25 =$	141,20 [kN]
Connector		
Material	EN AW 2011 (AlCuBiPb F37)	
Geometry	Dm=	37,8 [mm]
<u>Bearing stress in connector</u>	fu / YM2=	248,00 [N/mm ²]
	do=	16 [mm]
	t=	37,8 [mm]
	e1=	23,5 [mm]
	αb=	0,49 [-]
	e2=	16 [mm]
	k1=	1,1 [-]
	$NRd = k1 \times \alpha_b \times f_u \times d \times t / YM2 =$	80,78 [kN]
Remaining section under tension		
	$NRd = 0,9 \times A_{,net} \times f_u / YM2 =$	115,49 [kN]
Female Fitting		
Geometry	DH=	59 [mm]
	Di-1=	39,4 [mm]
	Di-2=	45,2 [mm]
	Di-m=	42,3 [mm]
<u>Bearing stress in female fitting</u>	fu / YM2=	232 [N/mm ²]
	do=	17,1 [mm]
	t = DH - Di-m=	16,7 [mm]
	e1>	29,4 [mm]
	αb=	0,57
	e2>	30 [mm]
	k1=	2,5
	$NRd = k1 \times \alpha_b \times f_u \times d \times t / YM2 =$	94,92 [kN]

The allowable normal force of the coupler is not relevant compared to the allowable normal force of the tube.
(NRd_G = 54,33 kN < 80,78 kN).



Interaction bending and normal force at coupler

Normal force and bending moments are transferred by the welding seam between coupler and chord.

Verification of interaction bending and normal force at coupler

$$\Rightarrow (N_{sdG} / NR_{dG})^{1,3} + (M_{sdG} / MR_{dG}) < 1,0$$

mit $N_{sdG} = N_{sd} / 4 + M_{sd} / (2 \cdot 0,47 \text{ m})$

und $M_{sdG} = 0,25 \cdot Q_{sd} \cdot 8,0 \text{ cm} = 2,0 \text{ cm} \cdot Q_{sd}$

$\Rightarrow a = \text{factor for cantilever at the coupler} = 2,0 \text{ cm}$

N_{sd} , M_{sd} und Q_{sd} : global internal forces in the truss (in kN resp. kNm)

The global internal forces include the following safety factors acc. Eurocode:

Selfweight of the truss: $\gamma_F = 1,35$

Net load on the truss: $\gamma_F = 1,50$

$NR_{dG} =$ allowable loading of the chord in the heat affected zone (see following table):

Main chord in heat affected zone at coupler			
$NR_d = A \times 0,8^* \times f_{u,haz} / \gamma_{M2} =$	68,44	[kN]	*(WIG TIG) local welding seam acc. chapter 6.2.9.3 (1)

$MR_{dG} = MuR_d$ (see following table):

Local bending of chord			
local welding seam acc. Chapter 6.2.9.3 (1)			
$D =$	50	[mm]	
red-Faktor =	0,8	[-]	(WIG TIG)
$\rho_{o,haz} =$	0,64	[-]	$f_{u,haz} / f_u$
$t_{u,eff} =$	2,04	[mm]	red-Faktor $\cdot \rho_{u,haz} \cdot t$
$W_{net} = \pi \times R^2 \times t_{u,eff} =$	3,39	[cm ³]	with $R = D / 2 - t / 2$
$MuR_d = W_{net} \cdot f_u / \gamma_{M2} =$	78,71	[kNcm]	acc. equation 6.24

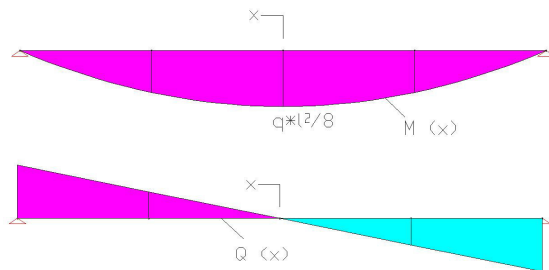


The following 3 cases are taken into account.

1. Loading with **uniformly distributed load (UDL)**

The coupler is always located at the theoretically worst point. This results from the following extremum-calculation:

Note: For the determination of the worst position of the coupler, the exponent 1,3, for the relation of actual load to the allowable load is not taken into account with sufficient accuracy.



$$M_{sd}(x) = q_{sd} \cdot L^2 / 8 - q_{sd} \cdot x^2 / 2$$

$$Q_{sd}(x) = q_{sd} \cdot x$$

$$NR_{dG} = 68,44 \text{ kN}$$

$$MR_{dG} = 78,71 \text{ kNcm}$$

$$\text{M-Q Interaction} \quad M_{sd}(x) / (2 \cdot 0,47 \cdot 68,44) + Q_{sd}(x) \cdot 2,0 / 78,71$$

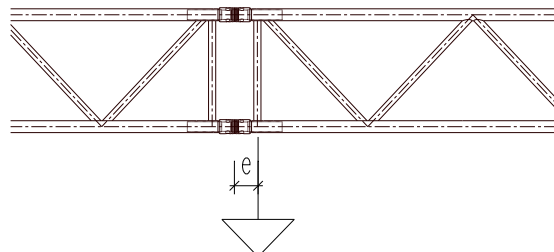
$$\text{Extremum:} \quad d/dx \sigma(x) = 0$$

$$\Rightarrow x = 2 \cdot 0,47 \cdot 68,44 \cdot 2 / 78,71 = 1,63 \text{ m}$$

(from middle of span)

2. Loading with **single-point loads** without requirements for position of couplers

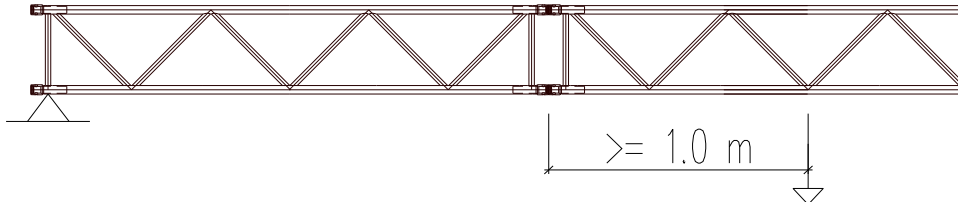
No requirements for position of coupler (Distance between load and coupler is e, see below). The allowable loads of this case have to be regarded, if the distance between coupler and load is <1,0 m.





3. Alternative to (2): Loading with single-point loads with requirements for position of couplers

In this case it is assumed, that the coupler is located $>1,0\text{m}$ from the loading-point. The coupler has to be located in that manner, so that the distance is $>1,0\text{m}$ into the direction of the support-point.



Summary

Following points are relevant for the determination of the allowable loads:

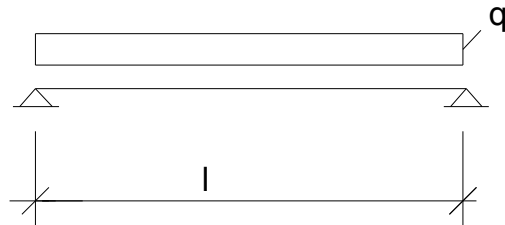
1. Allowable normal force in main chord (NR_{dG})
Main chord in heat affected zone at coupler is relevant $\Rightarrow NR_{dG} = 54,33 \text{ kN}$
2. Global shear force in truss (Q)
Allowable normal force in diagonals at nodes is relevant $\Rightarrow NR_{dD} = 30,13 \text{ kN}$
allowable shear force: $QR_d / (2 \cdot \sin 44,6^\circ) < 0,9 \cdot NR_{dD}$
(10% reduction because of minor stresses)
 $\Rightarrow \text{zul } QR_d = 0,9 \cdot 30,13 \cdot 2 \cdot \sin 44,6^\circ \Rightarrow QR_d = 38,08 \text{ kN}$
3. Interaction bending and normal force at coupler see pg. 14



5 ALLOWABLE LOADING SINGLE SPAN GIRDER

5.1 Uniformly distributed load (UDL)

System:



$$q_{sd} = p_{sd} + g_{sd}$$

(payload + selfweight, including safety)

Normal force in chord:

$$NR_d \geq q_{sd} \cdot L^2 / 8 / (n \cdot b)$$

$$\Rightarrow q_{sd} \leq NR_d \cdot (n \cdot b) \cdot 8 / L^2$$

$$\Rightarrow \text{zul } p = (NR_d \cdot (n \cdot b) \cdot 8 / L^2 - g_{sd}) / \gamma_F$$

Normal force in diagonals:

$$QR_d \geq q_{sd} \cdot L / 2$$

$$\Rightarrow q_{sd} \leq QR_d \cdot 2 / L$$

$$\Rightarrow \text{zul } p = (QR_d \cdot 2 / L - g_{sd}) / \gamma_F$$

Interaction at coupler:

Verification of interaction bending and normal force at coupler by an iterative method.

$$\Rightarrow (N_{sdG} / NR_{dG})^{1,3} + (M_{sdG} / MR_{dG}) < 1,0$$

applied:

The coupler is located at $e = 1,63$ m from the middle of the span (theoretically worst point, see extremum-calculation in chapter 4))

Loading-table:

see next page



Loading table:

The coupler is located at $e = 1,63$ m from the middle of the span
(theoretically worst point, see extremum-calculation in chapter 4)

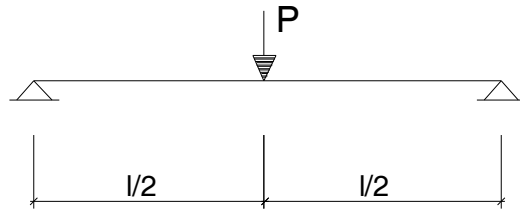
Uniformly distributed load UDL

	allowable load as a function of			
	Nrd	Qrd	Interaction at coupler	
L [m]	zul q [kN/m]	zul q [kN/m]	zul q [kN/m]	min zul q [kN/m]
4,00	16,88	12,55	13,73	12,55
5,00	10,75	10,01	10,39	10,01
6,00	7,42	8,32	7,83	7,42
7,00	5,41	7,11	6,01	5,41
8,00	4,11	6,20	4,72	4,11
9,00	3,22	5,50	3,78	3,22
10,00	2,58	4,93	3,08	2,58
11,00	2,11	4,47	2,55	2,11
12,00	1,75	4,09	2,14	1,75
13,00	1,47	3,76	1,82	1,47
14,00	1,25	3,48	1,55	1,25
15,00	1,07	3,24	1,34	1,07
16,00	0,92	3,03	1,17	0,92
17,00	0,80	2,84	1,02	0,80
18,00	0,70	2,68	0,90	0,70
19,00	0,61	2,53	0,79	0,61
20,00	0,54	2,39	0,70	0,54
21,00	0,47	2,27	0,62	0,47
22,00	0,42	2,16	0,56	0,42
23,00	0,37	2,06	0,50	0,37
24,00	0,33	1,97	0,45	0,33



5.2 Single-load in 1/2 point

System



Normal force in chord:

$$NRd \geq (P_{sd} \cdot L / 4 + g_{sd} \cdot L^2 / 8) / (n \cdot b)$$

$$\Rightarrow P_{sd} \leq [NRd \cdot (n \cdot b) - g_{sd} \cdot L^2 / 8] \cdot 4 / L$$

$$\Rightarrow \text{zul } P = [NRd \cdot (n \cdot b) - g_{sd} \cdot L^2 / 8] \cdot 4 / L / y_F$$

Normal force in diagonals:

$$QRd \geq P_{sd} / 2 + g_{sd} \cdot L / 2$$

$$\Rightarrow P_{sd} \leq (QRd - g_{sd} \cdot L / 2) \cdot 2$$

$$\Rightarrow \text{zul } P = (QRd - g_{sd} \cdot L / 2) \cdot 2 / y_F$$

Interaction at coupler:

Verification of interaction bending and normal force at coupler by an iterative method.

$$\Rightarrow (N_{sdG} / NR_{dG})^{1,3} + (M_{sdG} / MR_{dG}) < 1,0$$

applied:

Loading point at coupler

$e = 0,095 \text{ m}$

Alternatively: Location of coupler from loading point

$\geq 1,0 \text{ m}$

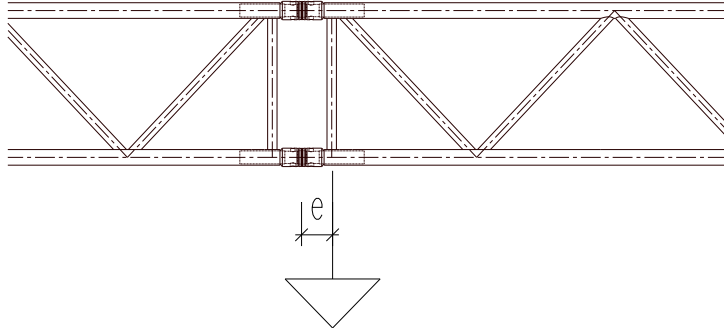
Loading-tables:

see next pages



Loading point at coupler

$e = 0,095 \text{ m}$

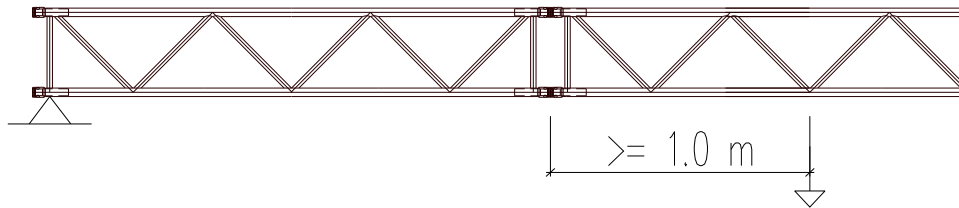


Single-load in 1/2point

L [m]	allowable load as a function of			min zul P [kN]
	Nrd	Qrd	Interaction at coupler	
			0,095	= e [m]
	zul P [kN]	zul P [kN]	zul P [kN]	
4,00	33,76	50,20	26,30	26,30
5,00	26,88	50,05	22,78	22,78
6,00	22,27	49,91	20,03	20,03
7,00	18,95	49,77	17,83	17,83
8,00	16,45	49,62	16,02	16,02
9,00	14,48	49,48	14,51	14,48
10,00	12,90	49,33	13,23	12,90
11,00	11,59	49,19	12,13	11,59
12,00	10,48	49,05	11,18	10,48
13,00	9,54	48,90	10,33	9,54
14,00	8,72	48,76	9,59	8,72
15,00	8,00	48,61	8,92	8,00
16,00	7,36	48,47	8,32	7,36
17,00	6,79	48,33	7,77	6,79
18,00	6,27	48,18	7,27	6,27
19,00	5,80	48,04	6,81	5,80
20,00	5,37	47,89	6,38	5,37
21,00	4,97	47,75	5,98	4,97
22,00	4,61	47,61	5,61	4,61
23,00	4,27	47,46	5,26	4,27
24,00	3,95	47,32	4,93	3,95



Location of coupler from loading point $e \geq 1,0$ m



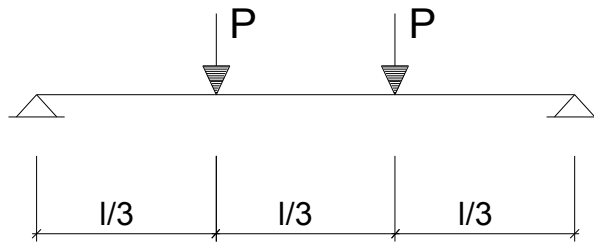
Single-load in 1/2point

L [m]	allowable load as a function of			
	Nrd	Qrd	Interaction at coupler	
	zul P [kN]	zul P [kN]	1	= e [m]
			zul P [kN]	min zul P [kN]
4,00	33,76	50,20	35,34	33,76
5,00	26,88	50,05	29,46	26,88
6,00	22,27	49,91	25,11	22,27
7,00	18,95	49,77	21,79	18,95
8,00	16,45	49,62	19,19	16,45
9,00	14,48	49,48	17,10	14,48
10,00	12,90	49,33	15,37	12,90
11,00	11,59	49,19	13,93	11,59
12,00	10,48	49,05	12,71	10,48
13,00	9,54	48,90	11,65	9,54
14,00	8,72	48,76	10,73	8,72
15,00	8,00	48,61	9,92	8,00
16,00	7,36	48,47	9,19	7,36
17,00	6,79	48,33	8,54	6,79
18,00	6,27	48,18	7,95	6,27
19,00	5,80	48,04	7,41	5,80
20,00	5,37	47,89	6,92	5,37
21,00	4,97	47,75	6,47	4,97
22,00	4,61	47,61	6,04	4,61
23,00	4,27	47,46	5,65	4,27
24,00	3,95	47,32	5,28	3,95



5.3 Single-loads in 1/3 points

System



Normal force in chord:

$$NRd \geq (P_{sd} \cdot L / 3 + g_{sd} \cdot L^2 / 8) / (n \cdot b)$$

$$\Rightarrow P_{sd} \leq [NRd \cdot (n \cdot b) - g_{sd} \cdot L^2 / 8] \cdot 3 / L$$

$$\Rightarrow \text{zul } P = [NRd \cdot (n \cdot b) - g_{sd} \cdot L^2 / 8] \cdot 3 / L / y_F$$

Normal force in diagonals:

$$QRd \geq P_{sd} / 2 + g_{sd} \cdot L / 2$$

$$\Rightarrow P_{sd} \leq (QRd - g_{sd} \cdot L / 2)$$

$$\Rightarrow \text{zul } P = (QRd - g_{sd} \cdot L / 2) / y_F$$

Interaction at coupler:

Verification of interaction bending and normal force at coupler by an iterative method.

$$\Rightarrow (N_{sdG} / NR_{dG})^{1,3} + (M_{sdG} / MR_{dG}) < 1,0$$

applied: Loading point at coupler e = 0,095 m

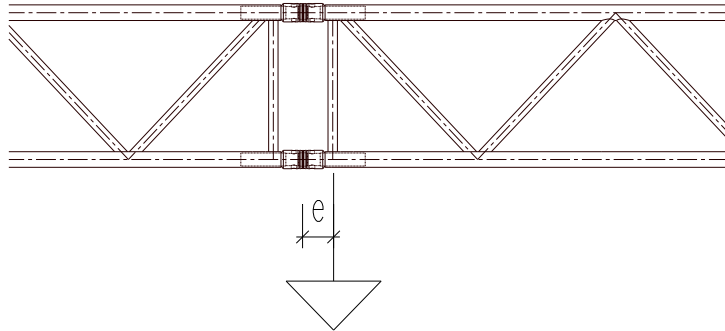
Alternatively: Location of coupler from loading point >= 1,0 m

Loading-tables: see next pages



Loading point at coupler

$e = 0,095 \text{ m}$



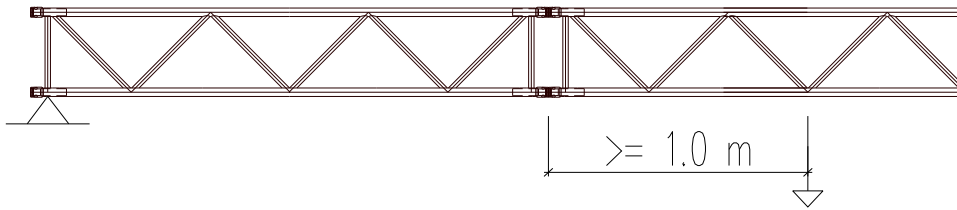
Single-load in 1/3points

L [m]	allowable load as a function of			
	Nrd	Qrd	Interaction at coupler	
	zul P [kN]	zul P [kN]	0,095	= e [m]
	zul P [kN]	zul P [kN]	zul P [kN]	min zul P [kN]
4,00	25,32	25,10	16,20	16,20
5,00	20,16	25,03	14,35	14,35
6,00	16,70	24,95	12,88	12,88
7,00	14,21	24,88	11,65	11,65
8,00	12,34	24,81	10,62	10,62
9,00	10,86	24,74	9,74	9,74
10,00	9,67	24,67	8,98	8,98
11,00	8,69	24,59	8,30	8,30
12,00	7,86	24,52	7,71	7,71
13,00	7,15	24,45	7,18	7,15
14,00	6,54	24,38	6,70	6,54
15,00	6,00	24,31	6,26	6,00
16,00	5,52	24,23	5,87	5,52
17,00	5,09	24,16	5,50	5,09
18,00	4,70	24,09	5,17	4,70
19,00	4,35	24,02	4,86	4,35
20,00	4,03	23,95	4,57	4,03
21,00	3,73	23,87	4,30	3,73
22,00	3,45	23,80	4,05	3,45
23,00	3,20	23,73	3,81	3,20
24,00	2,96	23,66	3,59	2,96



Location of coupler from loading point

$e \geq 1,0 \text{ m}$



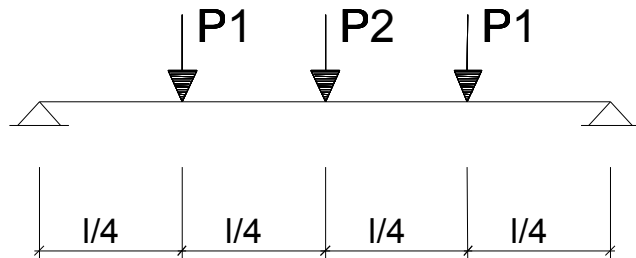
Single-load in 1/3points

L [m]	allowable load as a function of			
	Nrd	Qrd	Interaction at coupler	= e [m]
	zul P [kN]	zul P [kN]	1 zul P [kN]	
				min zul P [kN]
4,00	25,32	25,10	23,14	23,14
5,00	20,16	25,03	20,08	20,08
6,00	16,70	24,95	17,52	16,70
7,00	14,21	24,88	15,45	14,21
8,00	12,34	24,81	13,75	12,34
9,00	10,86	24,74	12,35	10,86
10,00	9,67	24,67	11,17	9,67
11,00	8,69	24,59	10,18	8,69
12,00	7,86	24,52	9,32	7,86
13,00	7,15	24,45	8,58	7,15
14,00	6,54	24,38	7,92	6,54
15,00	6,00	24,31	7,35	6,00
16,00	5,52	24,23	6,83	5,52
17,00	5,09	24,16	6,36	5,09
18,00	4,70	24,09	5,94	4,70
19,00	4,35	24,02	5,56	4,35
20,00	4,03	23,95	5,21	4,03
21,00	3,73	23,87	4,88	3,73
22,00	3,45	23,80	4,58	3,45
23,00	3,20	23,73	4,29	3,20
24,00	2,96	23,66	4,03	2,96



5.4 Single-loads in 1/4 points

System



Normal force in chord:

$$NRd \geq (Psd \cdot L / 4 + gsd \cdot L^2 / 8) / (n \cdot b)$$

$$\Rightarrow Psd \leq [NRd \cdot (n \cdot b) - gsd \cdot L^2 / 8] \cdot 2 / L$$

$$\Rightarrow \text{zul } P = [NRd \cdot (n \cdot b) - gsd \cdot L^2 / 8] \cdot 2 / L / yF$$

Normal force in diagonals:

$$QRd \geq 3 / 2 \cdot Psd + gsd \cdot L / 2$$

$$\Rightarrow Psd \leq (QRd - gsd \cdot L / 2) \cdot 2 / 3$$

$$\Rightarrow \text{zul } P = (QRd - gsd \cdot L / 2) \cdot 2 / 3 / yF$$

Interaction at coupler:

Verification of interaction bending and normal force at coupler by an iterative method.

$$\Rightarrow (Nsd_G / NRd_G)^{1.3} + (Msd_G / MRd_G) < 1,0$$

applied: Loading point at coupler e = 0,095 m

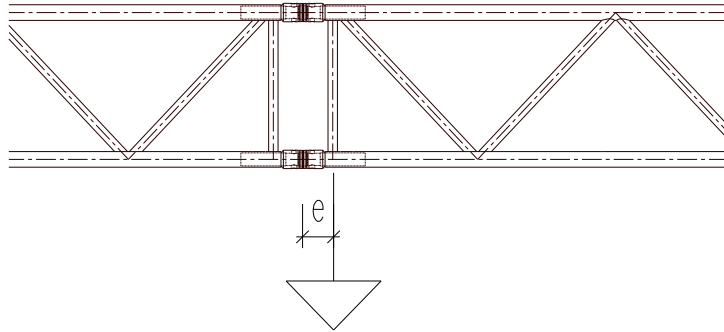
Alternatively: Location of coupler from loading point >= 1,0 m

Loading-tables: see next pages



Loading point at coupler

$e = 0,095 \text{ m}$



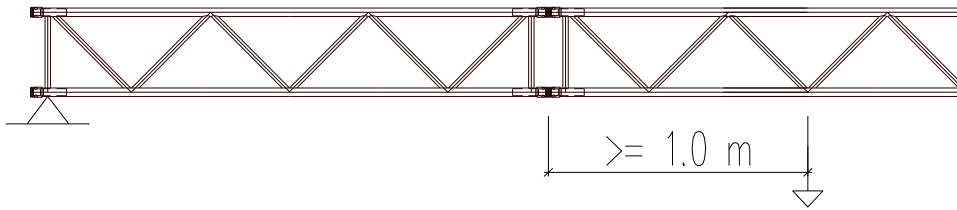
Single-load in 1/4points

L [m]	allowable load as a function of				
	Nrd	Qrd	Interaction at coupler 1	Interaction at coupler 2	
			0,095	0,095	= e [m]
zul P [kN]	zul P [kN]	zul P [kN]	zul P [kN]	zul P [kN]	min zul P [kN]
4,00	16,88	16,73	11,91	16,33	11,91
5,00	13,44	16,68	10,88	13,68	10,88
6,00	11,13	16,64	10,00	11,74	10,00
7,00	9,48	16,59	9,23	10,26	9,23
8,00	8,22	16,54	8,56	9,09	8,22
9,00	7,24	16,49	7,97	8,14	7,24
10,00	6,45	16,44	7,43	7,35	6,45
11,00	5,79	16,40	6,95	6,69	5,79
12,00	5,24	16,35	6,50	6,12	5,24
13,00	4,77	16,30	6,10	5,62	4,77
14,00	4,36	16,25	5,73	5,19	4,36
15,00	4,00	16,20	5,38	4,80	4,00
16,00	3,68	16,16	5,06	4,46	3,68
17,00	3,39	16,11	4,76	4,15	3,39
18,00	3,13	16,06	4,47	3,86	3,13
19,00	2,90	16,01	4,21	3,61	2,90
20,00	2,68	15,96	3,95	3,37	2,68
21,00	2,49	15,92	3,71	3,15	2,49
22,00	2,30	15,87	3,48	2,95	2,30
23,00	2,13	15,82	3,26	2,76	2,13
24,00	1,97	15,77	3,05	2,58	1,97



Location of coupler from loading point

$e \geq 1,0 \text{ m}$



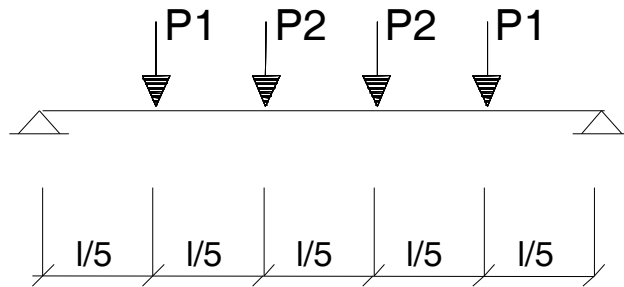
Single-load in 1/4points

L [m]	allowable load as a function of				
	Nrd	Qrd	Interaction at coupler 1	Interaction at coupler 2	= e [m]
	zul P [kN]	zul P [kN]	zul P [kN]	zul P [kN]	min zul P [kN]
4,00	16,88	16,73	17,29	19,67	16,73
5,00	13,44	16,68	15,93	15,96	13,44
6,00	11,13	16,64	14,40	13,39	11,13
7,00	9,48	16,59	13,03	11,51	9,48
8,00	8,22	16,54	11,83	10,06	8,22
9,00	7,24	16,49	10,79	8,91	7,24
10,00	6,45	16,44	9,88	7,98	6,45
11,00	5,79	16,40	9,08	7,21	5,79
12,00	5,24	16,35	8,38	6,55	5,24
13,00	4,77	16,30	7,75	5,99	4,77
14,00	4,36	16,25	7,18	5,50	4,36
15,00	4,00	16,20	6,67	5,07	4,00
16,00	3,68	16,16	6,20	4,69	3,68
17,00	3,39	16,11	5,77	4,35	3,39
18,00	3,13	16,06	5,38	4,05	3,13
19,00	2,90	16,01	5,01	3,77	2,90
20,00	2,68	15,96	4,67	3,51	2,68
21,00	2,49	15,92	4,35	3,28	2,49
22,00	2,30	15,87	4,06	3,06	2,30
23,00	2,13	15,82	3,78	2,86	2,13
24,00	1,97	15,77	3,51	2,67	1,97



5.5 Single-loads in 1/5 points

System



Normal force in chord:

$$NRd \geq (Psd \cdot 3/5 \cdot L + gsd \cdot L^2 / 8) / (n \cdot b)$$

$$\Rightarrow Psd \leq [NRd \cdot (n \cdot b) - gsd \cdot L^2 / 8] \cdot 5 / 3$$

$$\Rightarrow \text{zul } P = [NRd \cdot (n \cdot b) - gsd \cdot L^2 / 8] \cdot 5 / 3 / L / \gamma_F$$

Normal force in diagonals:

$$QRd \geq 2 \cdot Psd + gsd \cdot L / 2$$

$$\Rightarrow Psd \leq (QRd - gsd \cdot L / 2) / 2$$

$$\Rightarrow \text{zul } P = (QRd - gsd \cdot L / 2) / 2 / \gamma_F$$

Interaction at coupler:

Verification of interaction bending and normal force at coupler by an iterative method.

$$\Rightarrow (Nsd_G / NRd_G)^{1,3} + (Msd_G / MRd_G) < 1,0$$

applied:

Loading point at coupler

e = 0,095 m

Alternatively: Location of coupler from loading point

>= 1,0 m

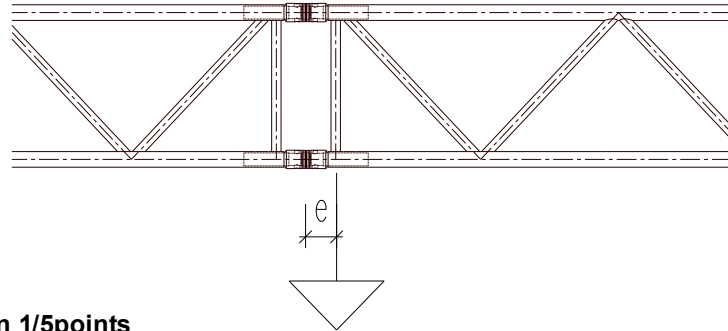
Loading-tables:

see next pages



Loading point at coupler

$e = 0,095 \text{ m}$



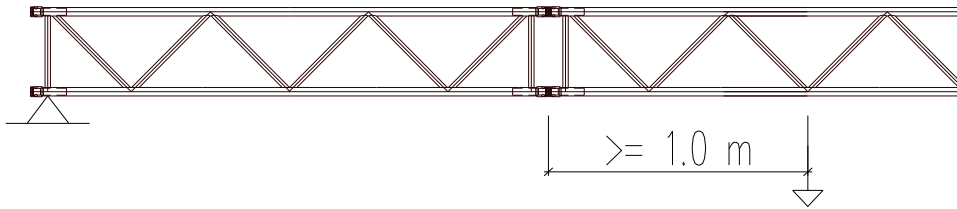
Single-load in 1/5points

allowable load as a function of					
	Nrd	Qrd	Interaction at coupler 1	Interaction at coupler 2	
			0,095	0,095	= e [m]
L [m]	zul P [kN]	zul P [kN]	zul P [kN]	zul P [kN]	min zul P [kN]
4,00	14,07	12,55	9,77	11,67	9,77
5,00	11,20	12,51	8,99	10,01	8,99
6,00	9,28	12,48	8,36	8,74	8,36
7,00	7,90	12,44	7,80	7,74	7,74
8,00	6,85	12,41	7,29	6,93	6,85
9,00	6,03	12,37	6,84	6,26	6,03
10,00	5,37	12,33	6,43	5,69	5,37
11,00	4,83	12,30	6,05	5,21	4,83
12,00	4,37	12,26	5,70	4,79	4,37
13,00	3,97	12,23	5,37	4,42	3,97
14,00	3,63	12,19	5,07	4,09	3,63
15,00	3,33	12,15	4,79	3,80	3,33
16,00	3,07	12,12	4,52	3,54	3,07
17,00	2,83	12,08	4,27	3,30	2,83
18,00	2,61	12,05	4,03	3,09	2,61
19,00	2,42	12,01	3,80	2,89	2,42
20,00	2,24	11,97	3,59	2,70	2,24
21,00	2,07	11,94	3,38	2,53	2,07
22,00	1,92	11,90	3,18	2,37	1,92
23,00	1,78	11,87	2,99	2,23	1,78
24,00	1,64	11,83	2,81	2,09	1,64



Location of coupler from loading point

$e \geq 1,0 \text{ m}$



Single-load in 1/5points

	allowable load as a function of				
	Nrd	Qrd	Interaction at coupler 1	Interaction at coupler 2	
			1	1	= e [m]
L [m]	zul P [kN]	zul P [kN]	zul P [kN]	zul P [kN]	min zul P [kN]
4,00	14,07	12,55	not relevant	not relevant	12,55
5,00	11,20	12,51	12,91	12,64	11,20
6,00	9,28	12,48	12,12	10,72	9,28
7,00	7,90	12,44	11,20	9,27	7,90
8,00	6,85	12,41	10,32	8,15	6,85
9,00	6,03	12,37	9,53	7,25	6,03
10,00	5,37	12,33	8,81	6,51	5,37
11,00	4,83	12,30	8,16	5,90	4,83
12,00	4,37	12,26	7,58	5,37	4,37
13,00	3,97	12,23	7,05	4,92	3,97
14,00	3,63	12,19	6,57	4,53	3,63
15,00	3,33	12,15	6,13	4,18	3,33
16,00	3,07	12,12	5,72	3,88	3,07
17,00	2,83	12,08	5,35	3,60	2,83
18,00	2,61	12,05	5,00	3,35	2,61
19,00	2,42	12,01	4,68	3,12	2,42
20,00	2,24	11,97	4,37	2,91	2,24
21,00	2,07	11,94	4,09	2,72	2,07
22,00	1,92	11,90	3,82	2,55	1,92
23,00	1,78	11,87	3,57	2,38	1,78
24,00	1,64	11,83	3,33	2,23	1,64



6 SUMMARY OF THE RESULTS

6.1 Allowable loadings

Allowable load F54S

single-span beam

Span	UDL	Single point loads			
		in 1/2 Point	in 1/3 Points	in 1/4 Points	in 1/5 Points
[m]	[kN/m]	[kN]	[kN]	[kN]	[kN]
4	12,55	26,30	16,20	11,91	9,77
5	10,01	22,78	14,35	10,88	8,99
6	7,42	20,03	12,88	10,00	8,36
7	5,41	17,83	11,65	9,23	7,74
8	4,11	16,02	10,62	8,22	6,85
9	3,22	14,48	9,74	7,24	6,03
10	2,58	12,90	8,98	6,45	5,37
11	2,11	11,59	8,30	5,79	4,83
12	1,75	10,48	7,71	5,24	4,37
13	1,47	9,54	7,15	4,77	3,97
14	1,25	8,72	6,54	4,36	3,63
15	1,07	8,00	6,00	4,00	3,33
16	0,92	7,36	5,52	3,68	3,07
17	0,80	6,79	5,09	3,39	2,83
18	0,70	6,27	4,70	3,13	2,61
19	0,61	5,80	4,35	2,90	2,42
20	0,54	5,37	4,03	2,68	2,24
21	0,47	4,97	3,73	2,49	2,07
22	0,42	4,61	3,45	2,30	1,92
23	0,37	4,27	3,20	2,13	1,78
24	0,33	3,95	2,96	1,97	1,64

The values of the table are only valid for single-span girder.

The truss-elements have to be braced with diagonals.

Large loads have to be applied at the nodes or have to be distributed by appropriate constructions.

Loads at the middle of the couplers are not allowed.

All loads have to be distributed equally to both chords.



The specified values include partial safety coefficients on the loadings side acc. EN 1990 of $\gamma_F = 1.50$ for payloads and $\gamma_G = 1.35$ for selfweight of the truss.

For applications which can be calculated on the basis of other codes, the partial safety factors can be adjusted (for example temporary structures acc. EN 13814, $\gamma_F = 1.35$ for payloads).

To use the resulting allowable loads with British Standard (BS) and ANSI, allowable loads listed in tables have to be multiplied by 0.85.

The values are calculated with no requirements for the location of the couplers. In case that the distance from the couplers to the loadingpoints of the single-point loads is $\geq 1,0$ m into the direction of the support, the values of the calculations for the different loadcases can be used (see chapter 5).

6.2 Deflections at max. allowable loadings:

Deflections [cm] for F54 at max. allowable loads

[cm]

Span [m]	UDL [cm]	Single point loads			
		in 1/2 Point [cm]	in 1/3 Points [cm]	in 1/4 Points [cm]	in 1/5 Points [cm]
4	0,47	0,40	0,42	0,43	0,44
5	0,92	0,68	0,72	0,76	0,80
6	1,42	1,03	1,13	1,22	1,30
7	1,94	1,47	1,64	1,80	1,92
8	2,54	2,00	2,24	2,41	2,56
9	3,21	2,60	2,96	3,06	3,24
10	3,97	3,22	3,78	3,78	4,00
11	4,81	3,92	4,71	4,59	4,85
12	5,73	4,68	5,75	5,47	5,78
13	6,74	5,52	6,87	6,44	6,79
14	7,83	6,44	7,98	7,48	7,89
15	9,00	7,44	9,18	8,61	9,07
16	10,26	8,51	10,45	9,82	10,33
17	11,61	9,67	11,82	11,12	11,68
18	13,04	10,92	13,27	12,51	13,12
19	14,56	12,25	14,81	13,98	14,65
20	16,17	13,67	16,44	15,54	16,27
21	17,86	15,19	18,16	17,20	17,97
22	19,66	16,81	19,97	18,94	19,77
23	21,54	18,53	21,87	20,79	21,66
24	23,51	20,35	23,87	22,72	23,64

 = deflection > L/100



ANNEXE A: COMPARATIVE CALCULATION INFLUENCE OF HORIZONTAL LEVELS

The horizontal levels of the truss F54S are not stiffened by diagonal bracings. There are only vertical struts. By the following calculation it is examined to what extent this has an influence on the load bearing capacity of the truss.

For the calculation a system with a total length of 24m is chosen. The permissible loads for the F54S-Truss with $l = 24$ m are applied in the system.

Load cases:

LC1) Permissible uniformly distributed load for $l = 24$ m + self weight of the truss

$$p_v = (0,33 + 0,16) / 4 = 0,1225 \text{ kN/m per chord}$$

LC2) as LC1 + horizontal load of $p/100$

$$p_h = 0,135 / 100 = 0,001225 \text{ kN/m}$$

LC3) Calculation acc. second order theory with safety 1,5 acc. EN 1990

Results

(see software print)

acc. first order theory: LF2/LC2: $N_{\text{Chord}} = -37,12$ kN

acc. second order theory: LF3/LC3: $N_{\text{Chord}} = -55,81$ kN (including safety 1,5)

$$\Rightarrow N_{II} / N_I = 55,81 / 1,5 / 37,12 = 1,002 < 1,1$$

Increase of the internal forces resulting from calculation acc. second order theory resp. the influence of the deformation is negligible.

The remaining load reserve is determined by a iterative calculation: the factor of the load in the calculation according second order theory is increased until failure.

Load reserve UDL LC1: 2,92



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Annexe A

M 1 :

System characteristics

261 Nodes	
510 Elements	510 Beams
4 Supports	0 Slabs
0 Link elements	0 Plains
3 Material properties	0 Shells
3 Section properties	0 Cables
4 Load cases	0 Solids
0 LC Combinations	0 Spring elements
0 Tendon groups	

Result location in area elements: Node
5 Result locations in beam elements

Rotated element systems

0 Element systems
0 Internal force systems
0 Reinforcement systems

Section properties

1	<p>Polygon</p>	<p>Gurt</p> <p>Centroid [m] $y_s = -0,000$ $z_s = -0,000$ Area [m²] $A = 5,7435e-04$ Moments of inertia [m⁴] $I_x = 3,0395e-07$ $I_y = 1,5208e-07$ $I_1 = 1,5208e-07$ $I_z = 1,5208e-07$ $I_2 = 1,5208e-07$ Main axis angle [Grad] $\Phi = 0,000$ $I_{yz} = 0,0000e+00$ Averaging of the lateral force shear stress over section width</p>
2	<p>Polygon</p>	<p>Diagonale</p> <p>Centroid [m] $y_s = -0,000$ $z_s = -0,000$ Area [m²] $A = 2,5284e-04$ Moments of inertia [m⁴] $I_x = 4,6324e-08$ $I_y = 2,3175e-08$ $I_1 = 2,3175e-08$ $I_z = 2,3175e-08$ $I_2 = 2,3175e-08$ Main axis angle [Grad] $\Phi = 0,000$ $I_{yz} = 0,0000e+00$ Averaging of the lateral force shear stress over section width</p>
3	<p>Polygon</p>	<p>Querrohr</p> <p>Centroid [m] $y_s = -0,000$ $z_s = -0,000$ Area [m²] $A = 5,7435e-04$ Moments of inertia [m⁴] $I_x = 3,0395e-07$ $I_y = 1,5208e-07$ $I_1 = 1,5208e-07$ $I_z = 1,5208e-07$ $I_2 = 1,5208e-07$ Main axis angle [Grad] $\Phi = 0,000$ $I_{yz} = 0,0000e+00$ Averaging of the lateral force shear stress over section width</p>



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Annexe A

M 1 :

Material properties

	No.	Type	E-Modu. [MN/m ²]	G-Modu. [MN/m ²]	Poiss. ratio	alpha.t [1/K]	gamma [kN/m ³]	Miscellaneous
1	1	Frei	70000	29167	0,20	1,00e-05	27,000	fc = 145 [MN/m ²] ft = 145
2	2	Frei	70000	29167	0,20	1,00e-05	27,000	fc = 145 [MN/m ²] ft = 145
3	3	Frei	70000	29167	0,20	1,00e-05	27,000	fc = 145 [MN/m ²] ft = 145

List of load cases

LC.	Label
1	[Unnamed]
2	[Unnamed]
3	[Unnamed]

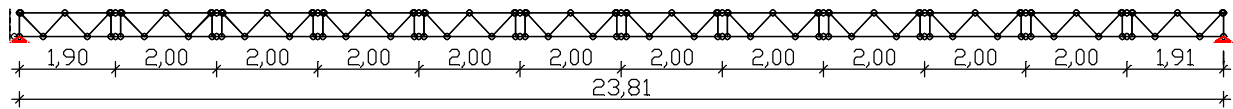
Sum of installed loads and support reactions

LC.	Label	Fx [kN]	Fy [kN]	Fz [kN]
1	[Unnamed]	-0,000	-0,000	11,679
	Support reactions	-0,000	-0,000	11,679
2	[Unnamed]	0,117	-0,000	11,679
	Support reactions	0,117	-0,000	11,679
3	[Unnamed]	0,175	-0,000	17,518
	Support reactions	0,175	0,000	17,518

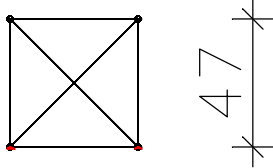


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Annexe A

30.06.2017
M 1 :



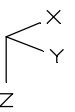
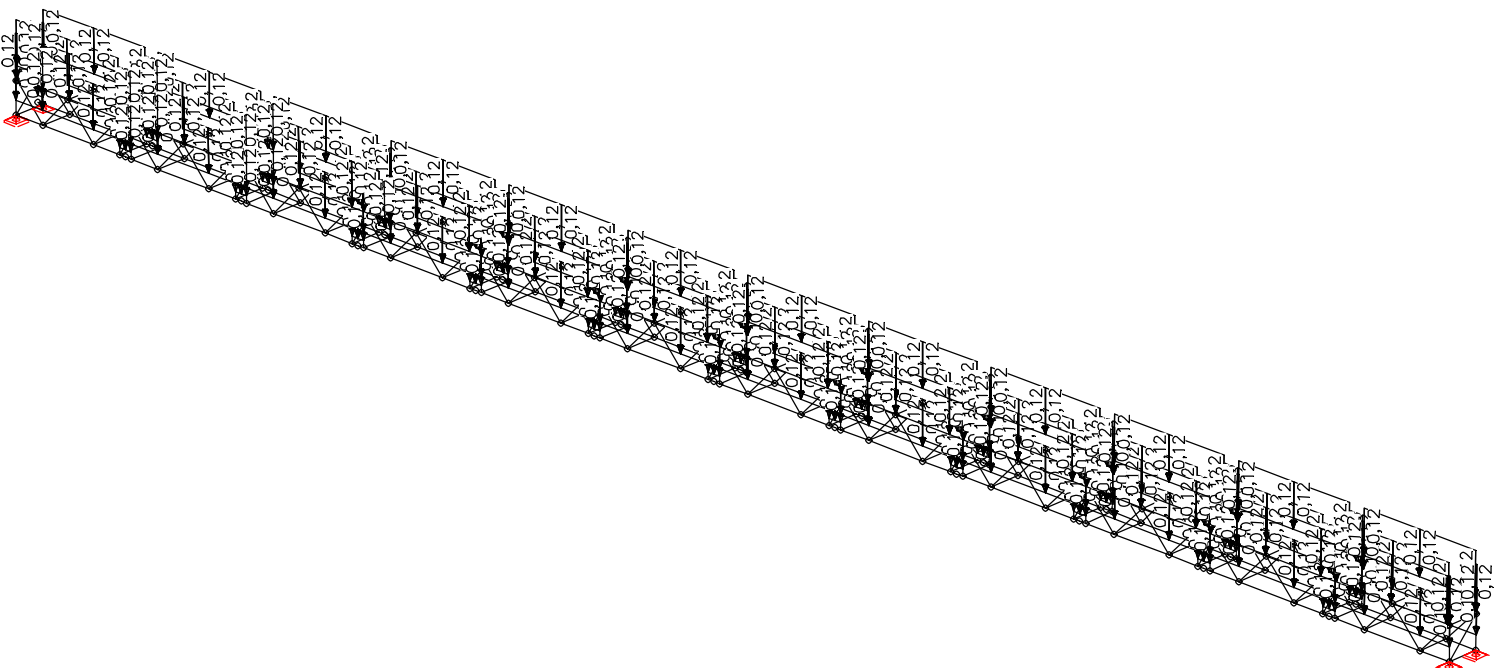
Dimensions



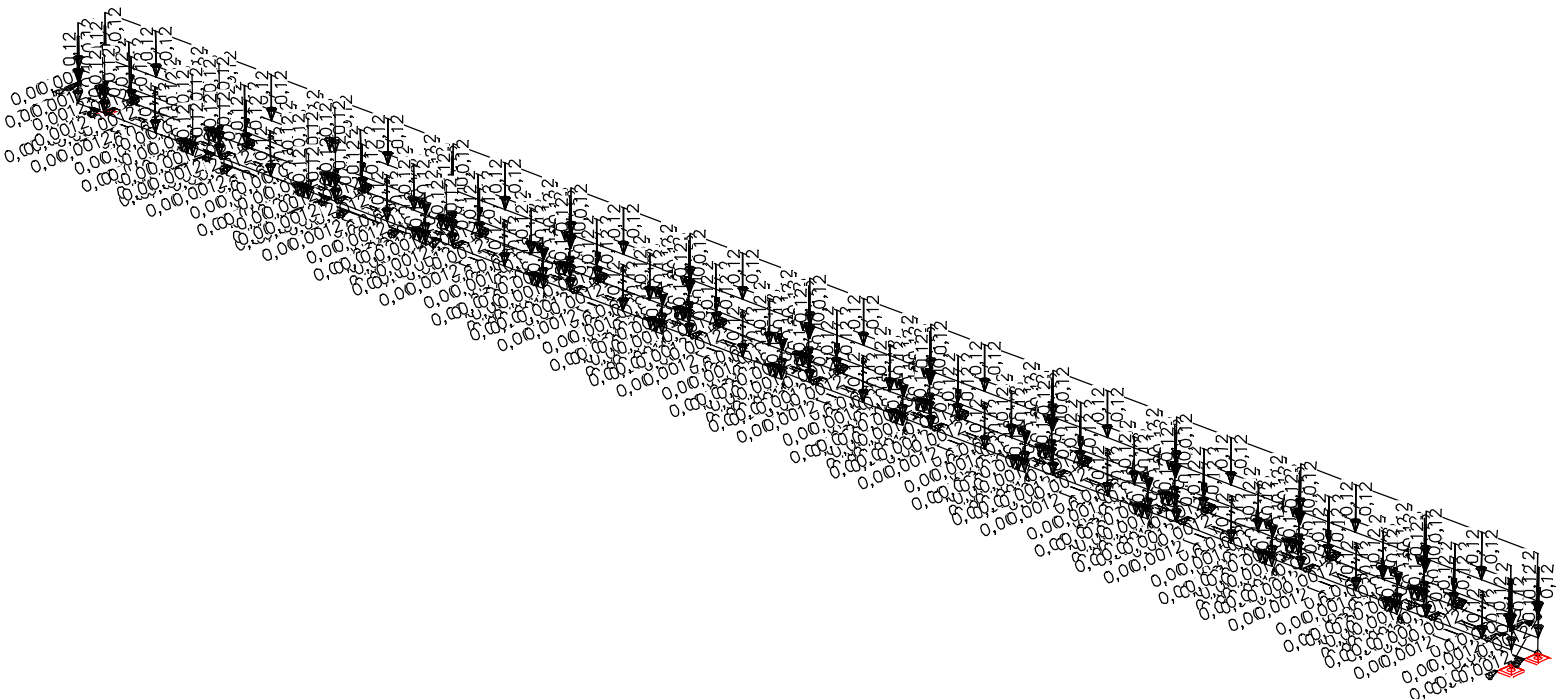
Dimensions



BUCKLING EIGENVALUES: n=4



LC 1: Load



LC 2: Load



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Annexe A

M 1 :

Load data load case 3:

Load group (GRL)

Theory: 2. order theory

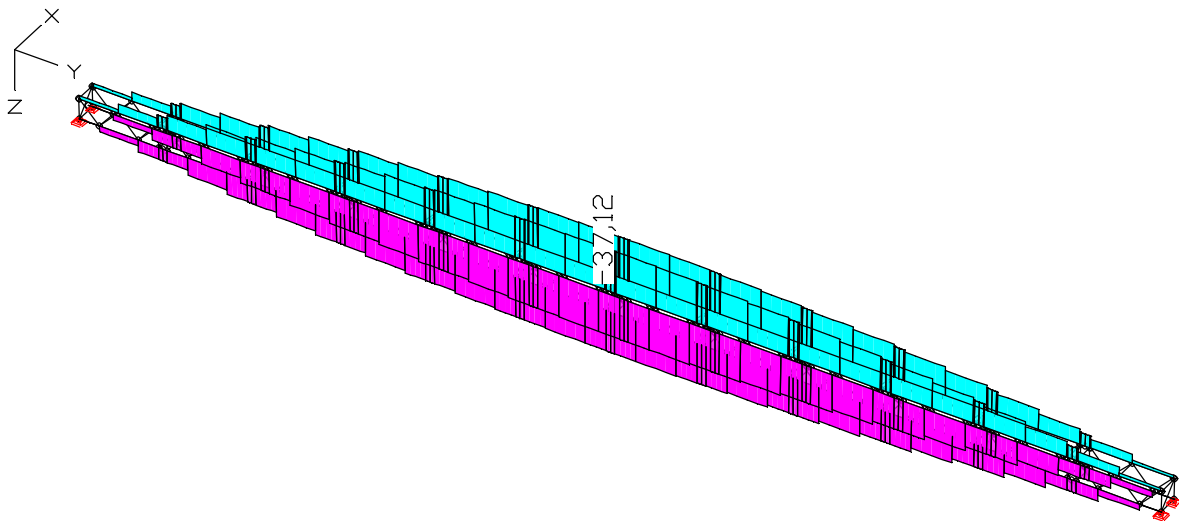
No soil pressure > 0: No; No support reac. < 0: No; Error threshold [%]: 1,00

Additional global load factor: 1,00; Predeformation: 0

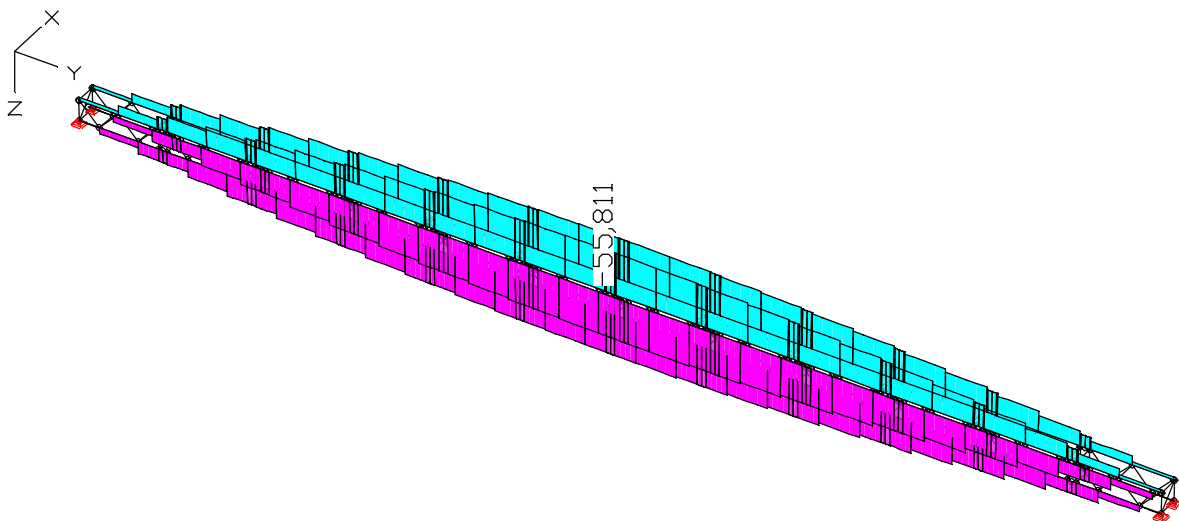
Consider concrete creeping in the nonlinear analysis: No

Selected load cases

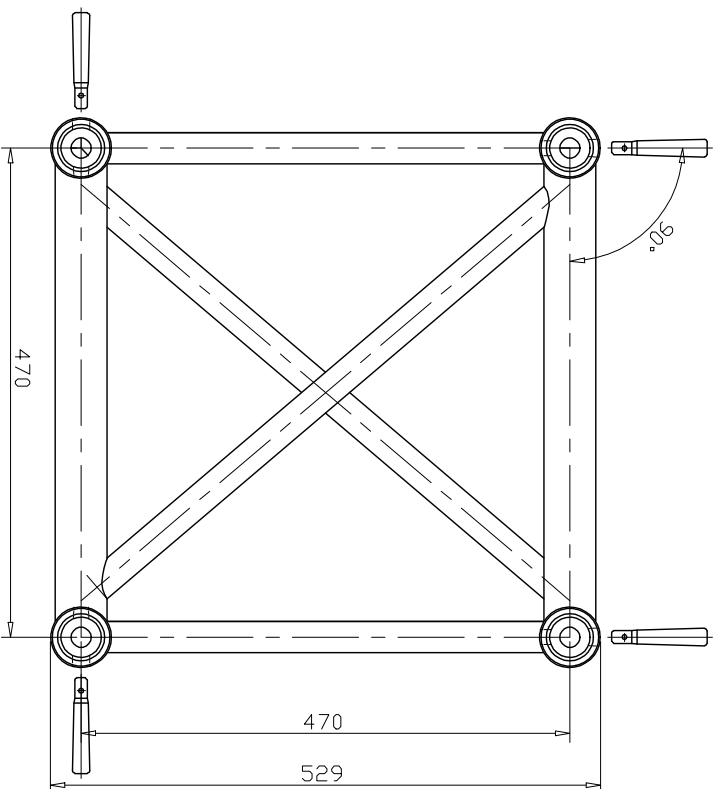
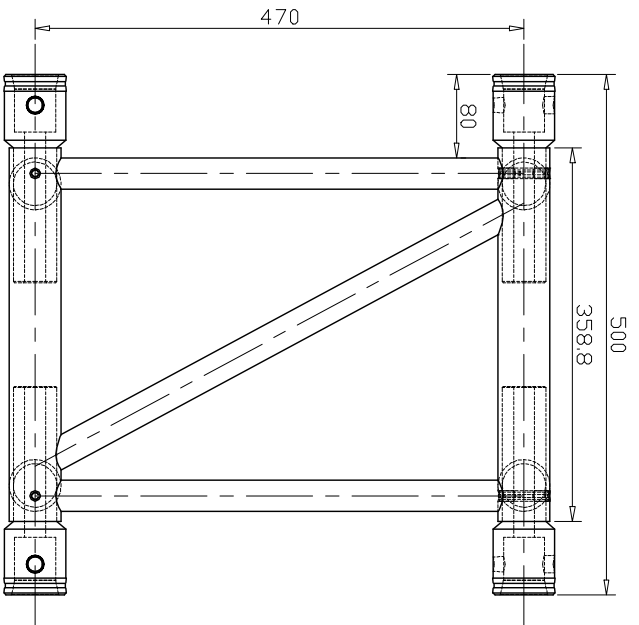
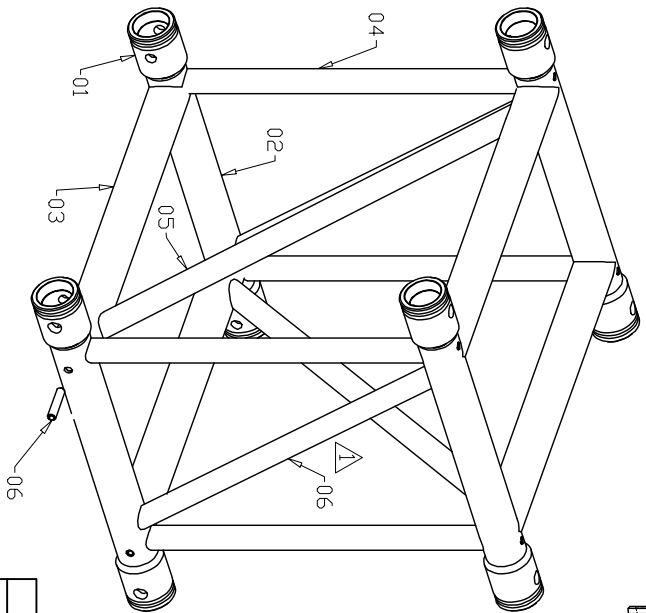
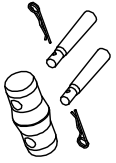
No.	Label	Factor
2		1,5



Internal forces Nx; LF 2



Internal forces Nx; LF 3

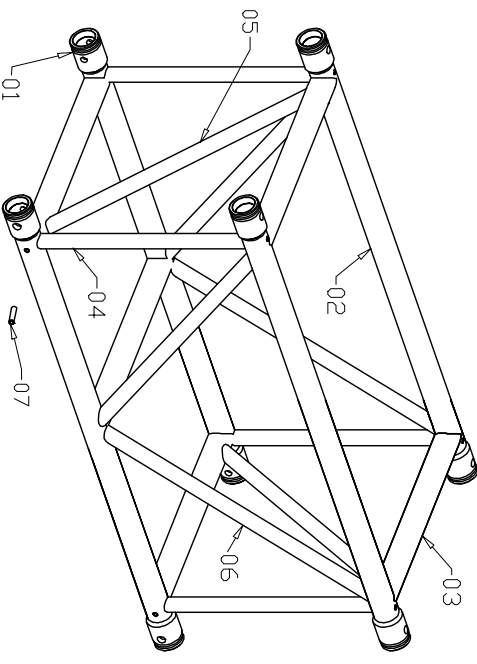
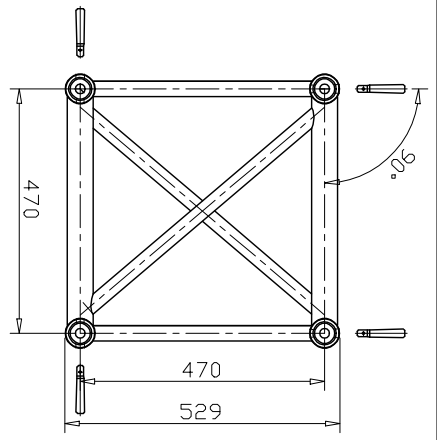
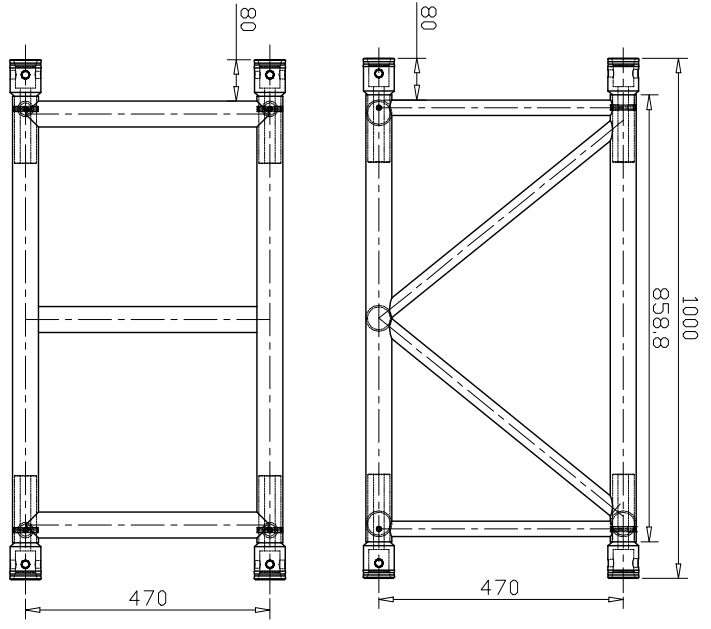
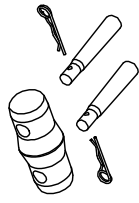


fitting	DMS-01-1	Steel	CP	8
	ST-F64-PJ-A-3	S45C	CP	8
	ST-F64-PJ-A-2	#2011	T6	4
	Ø10*50mm	Steel	Black	8
	Ø30*3.0t*493mm	#6082	T6	2
	Ø30*3.0t*577mm	#6082	T6	2
	Ø30*3.0t*419.5mm	#6082	T6	4
	Ø50*4.0t*454mm	#6082	T6	4
	Ø50*4.0t*358.8mm	#6082	T6	4
	ST-F64-PJ-A-1T	#6082	T6	8
ND.	SPEC.	MATERIAL	FIS-	QTY

DRAWN	F54S	DRW NO	F54S050	2D/3D	edition	1	
Specification	F54S-05M	DESIGN	Shendaulo	DATE	09.06.26	CHECKED	APPROVED
Weight	**kgs/lbs	UNITS	mm	DATE	Proofread		
SURFACE	RAW	SCALE	1:7				



MARK NAME	RETRIEVE CONTENTS	DATE

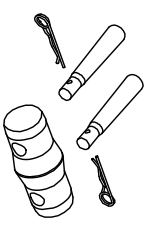
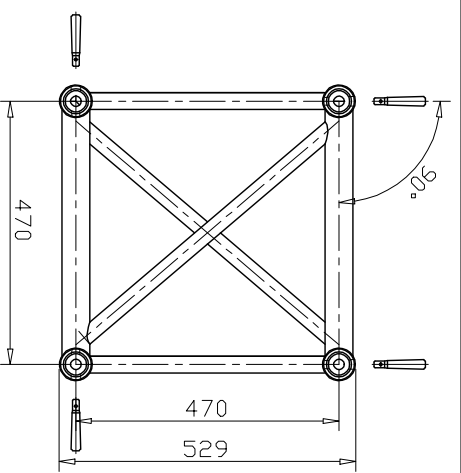
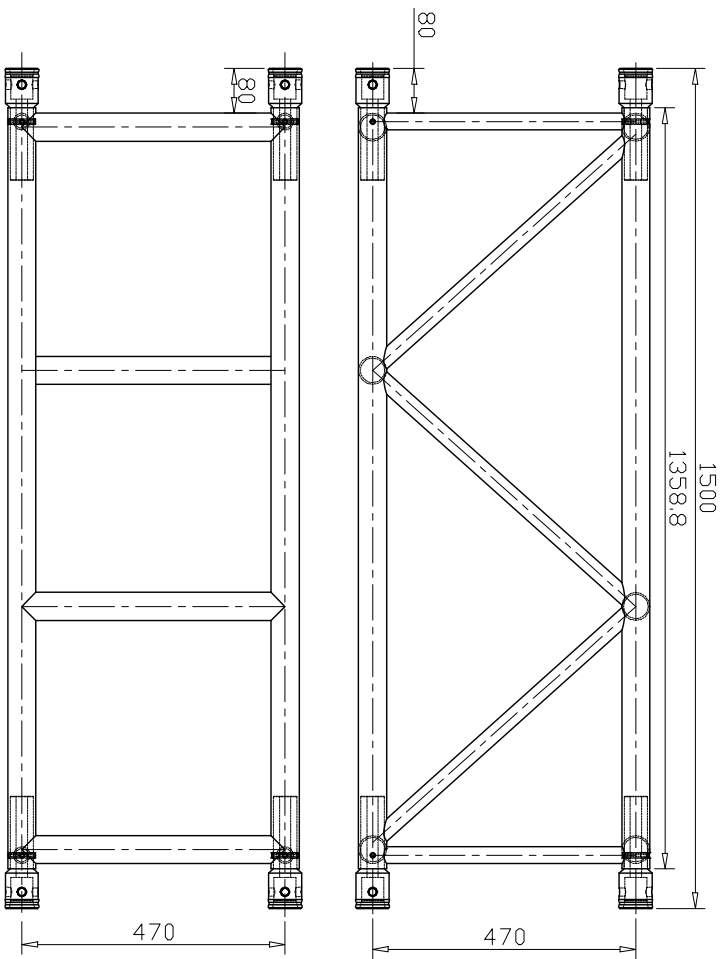


fitting	DMS-01-1	Steel	CP	8
	ST-F64-PJ-A-3	S45C	CP	8
	ST-F64-PJ-A-2	#2011	T6	4
	Ø10*50mm	Steel	Black	8
	Ø30*3.0t*565mm	#6082	T6	4
	Ø30*3.0t*577mm	#6082	T6	2
	Ø30*3.0t*419.5mm	#6082	T6	4
	Ø50*4.0t*454mm	#6082	T6	5
	Ø50*4.0t*858.8mm	#6082	T6	4
	ST-F64-PJ-A-1T	#6082	T6	8
ND.	SPEC.	MATERIAL	FIS-	QTY

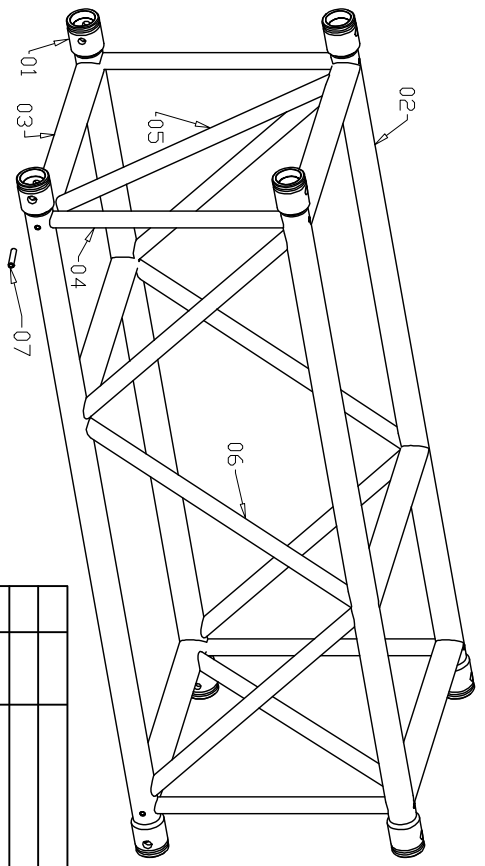
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Weight	**kgs/lbs	UNITS	mm	DATE	Proofread	APPROVED
SURFACE	RAW	SCALE	1:14			



MARK NAME	RETRIEVE CONTENTS	DATE



fitting

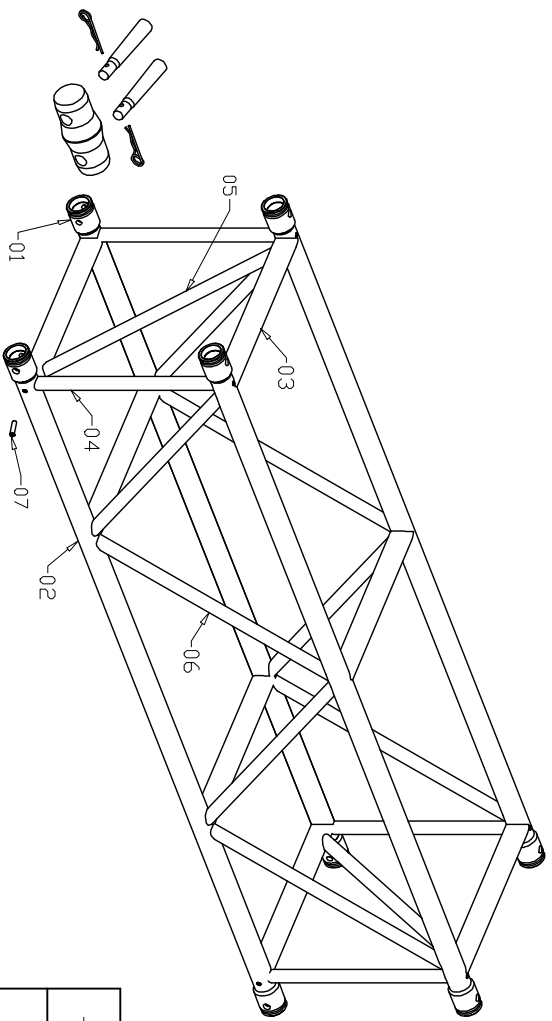
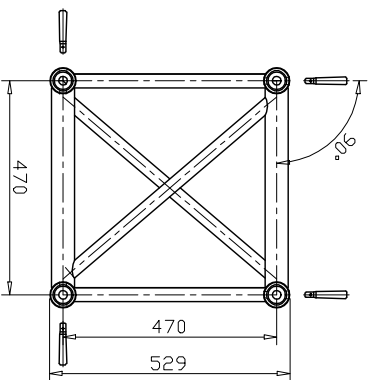
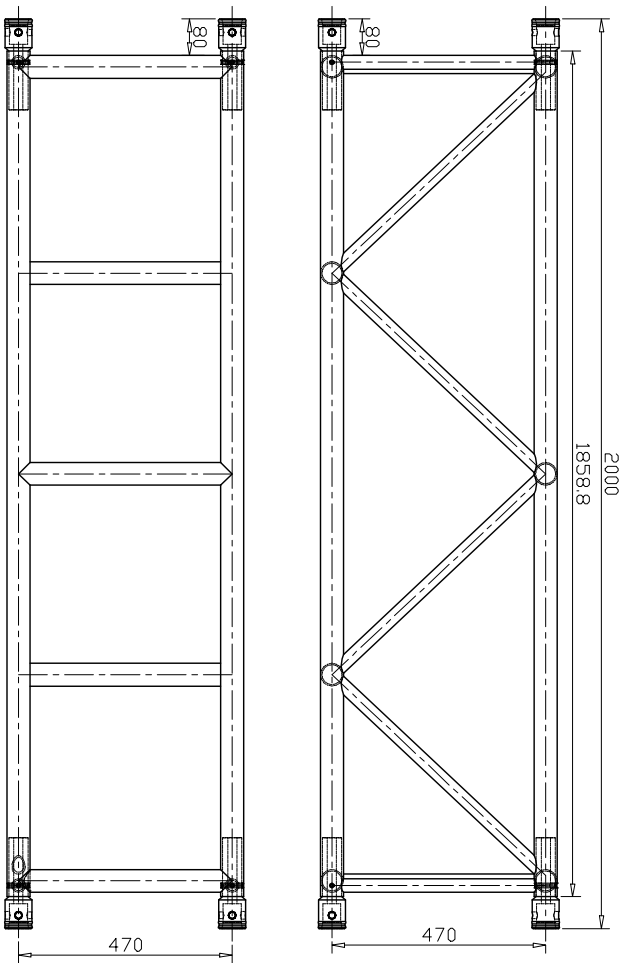


fitting	DMS-01-1	Steel	CP	8
	ST-F64-PJ-A-3	S45C	CP	8
	ST-F64-PJ-A-2	#2011	T6	4
	Ø10*50mm	Steel	Black	8
	Ø30*3.0t*591.1mm	#6082	T6	6
	Ø30*3.0t*577mm	#6082	T6	2
	Ø30*3.0t*419.5mm	#6082	T6	4
	Ø50*4.0t*454mm	#6082	T6	6
	Ø50*4.0t*1358.8mm	#6082	T6	4
	ST-F64-PJ-A-1T	#6082	T6	8
ND.	SPEC.	MATERIAL	FIS-	QTY

DRAWN	F54S	DRAW NO	F54S150	2D/3D	edition	1
Specification	F54S-1.5M	DESIGN	HUANGYUWEI	CHECKED	APPROVED	
Weight	**kgs/lbs	UNITS	mm	DATE	130131	
SURFACE	RAW	SCALE	1:1.3	Proofread		



MARK NAME	RETRIEVE CONTENTS	DATE

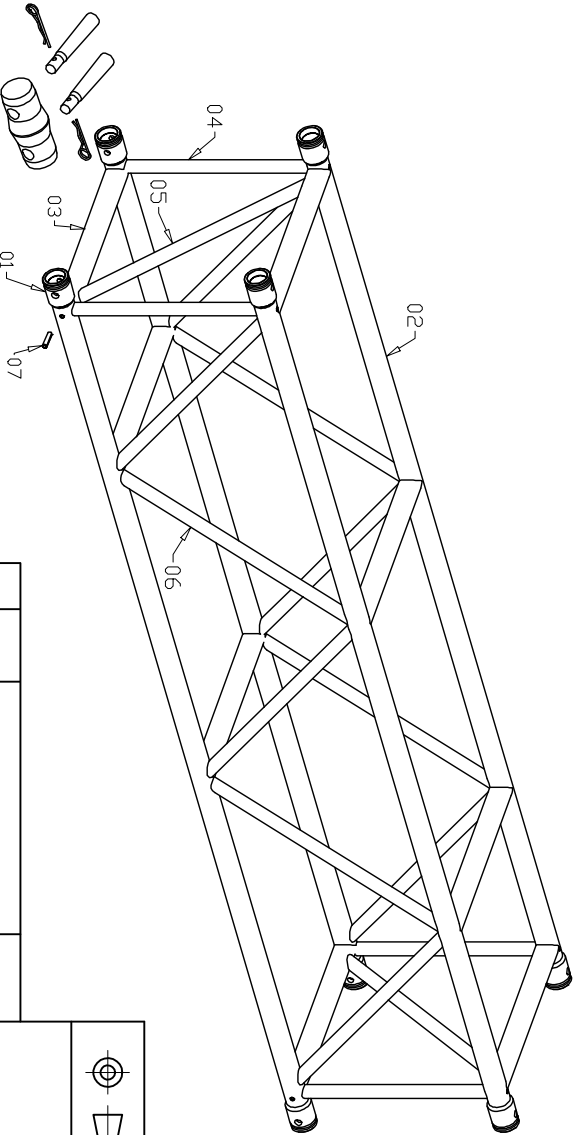
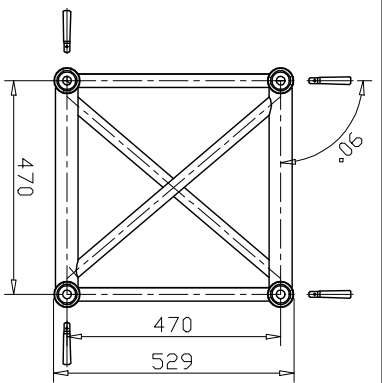
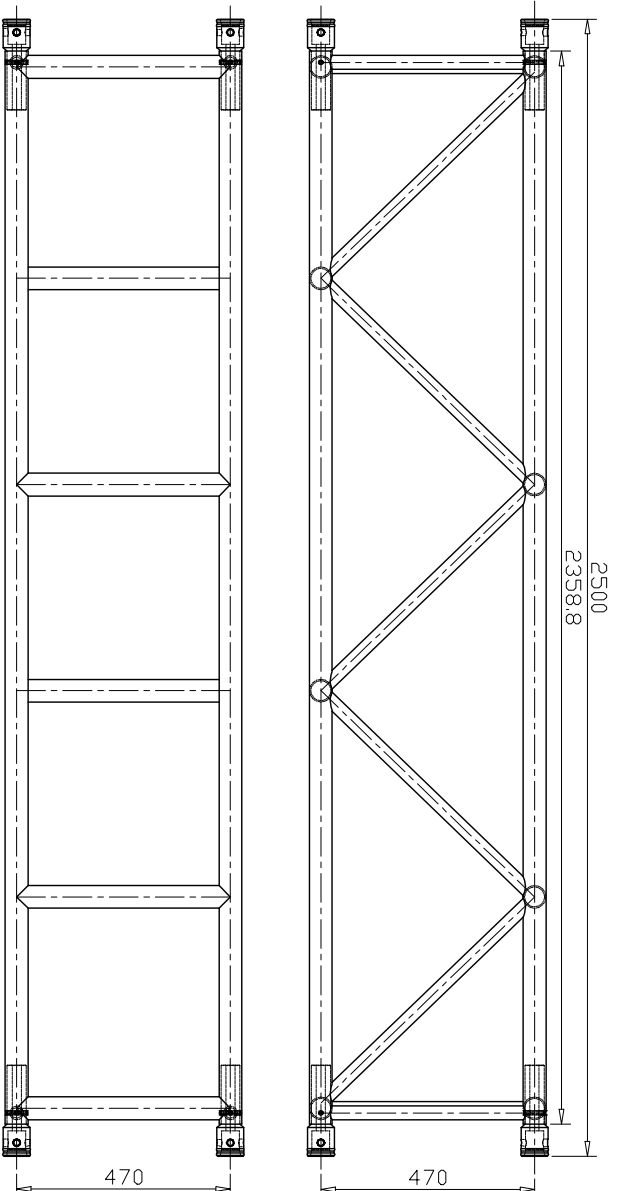


fitting	DMS-01-1	Steel	CP	8
	ST-F64-PJ-A-3	S45C	CP	8
	#2011	T6	T6	4
	ST-F64-PJ-A-2	Steel	Black	8
	Ø10*50mm			
	Ø30*3.0t*604.4mm	#6082	T6	8
	Ø30*3.0t*577mm	#6082	T6	2
	Ø30*3.0t*419.5mm	#6082	T6	4
	Ø30*3.0t*419.5mm	#6082	T6	4
	Ø50*4.0t*454mm	#6082	T6	7
	Ø50*4.0t*1858.8mm	#6082	T6	4
	ST-F64-PJ-A-1T	#6082	T6	8
	ND.	SPEC.	MATERIAL	FIS-QTY

DRAWN	F54S	DRAW NO	F54S200	2D/3D	edition	1
Specification	F54S-2.0M	DESIGN	Shendao	DATE	09.06.26	CHECKED
Weight	**kgs/lbs	UNITS	mm	DATE	Proofread	APPROVED
SURFACE	RAW	SCALE	1:16			



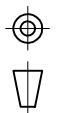
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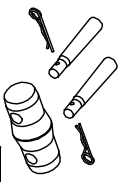
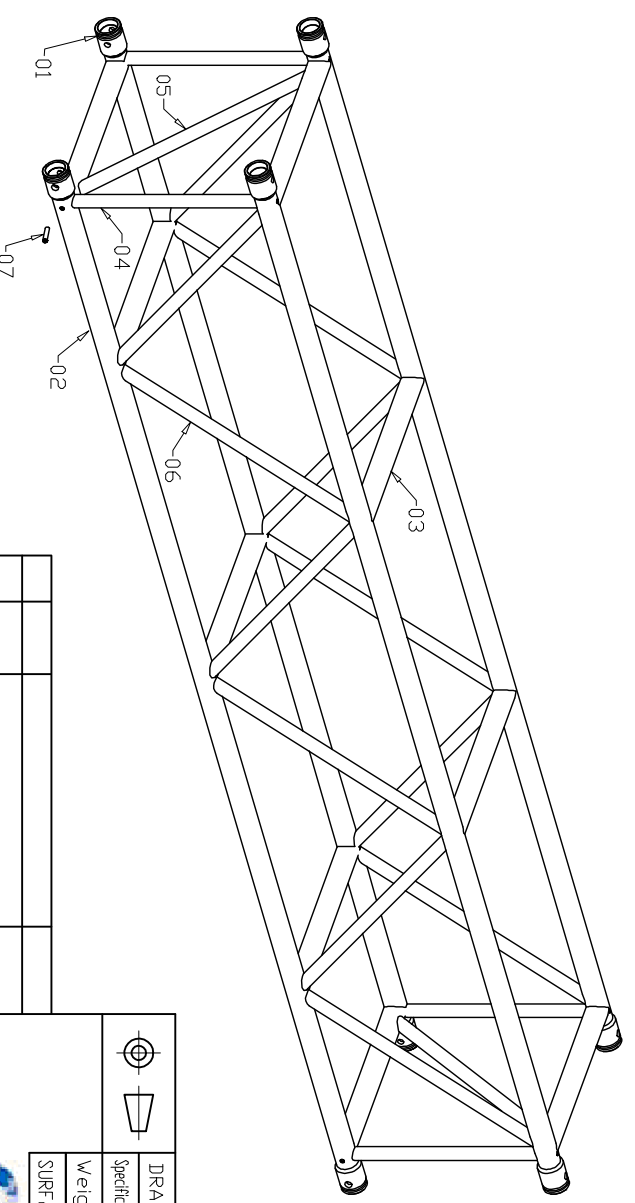
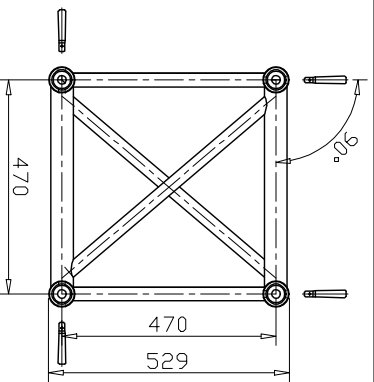
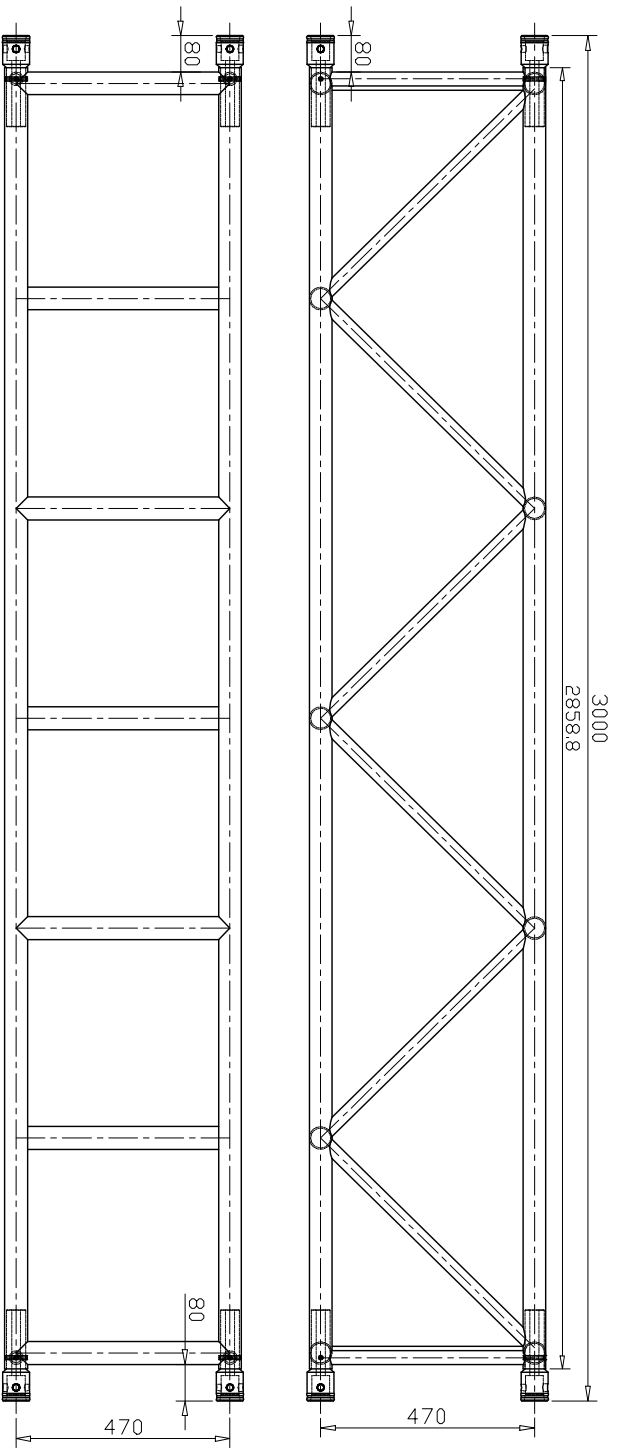


MARK NAME	RETRIEVE CONTENTS	DATE

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Specification	F54S-2.5M	DESIGN	Shendauloo	DATE	09.06.26
Weight	**kgs/lbs	UNITS	mm	CHECKED	
SURFACE	RAW	SCALE	1:16	APPROVED	

fitting	DMS-01-1	Steel	CP	8
ST-F64-PJ-A-3	S45C	CP	8	8
ST-F64-PJ-A-2	#2011	T6	4	4
Ø10*50mm	Steel	Black	8	8
Ø30*3.0t*612.5mm	#6082	T6	10	10
Ø30*3.0t*577mm	#6082	T6	2	2
Ø30*3.0t*419.5mm	#6082	T6	4	4
Ø50*4.0t*454mm	#6082	T6	8	8
Ø50*4.0t*2358.8mm	#6082	T6	4	4
ST-F64-PJ-A-1T	#6082	T6	8	8
ND.	SPEC.	MATERIAL	FIS-	QTY



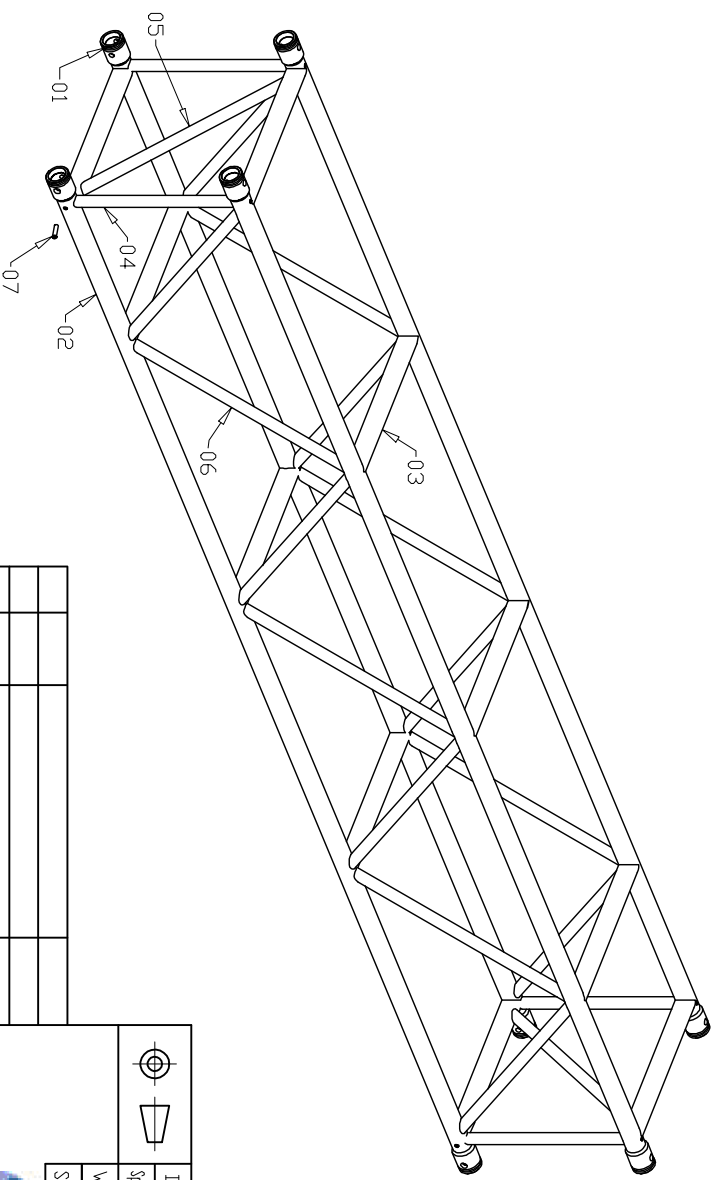
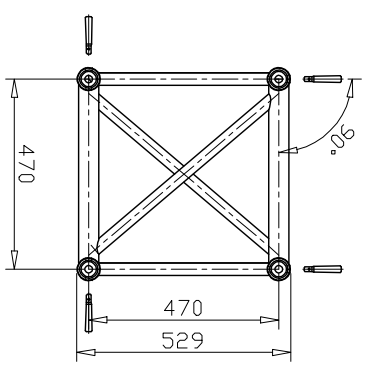
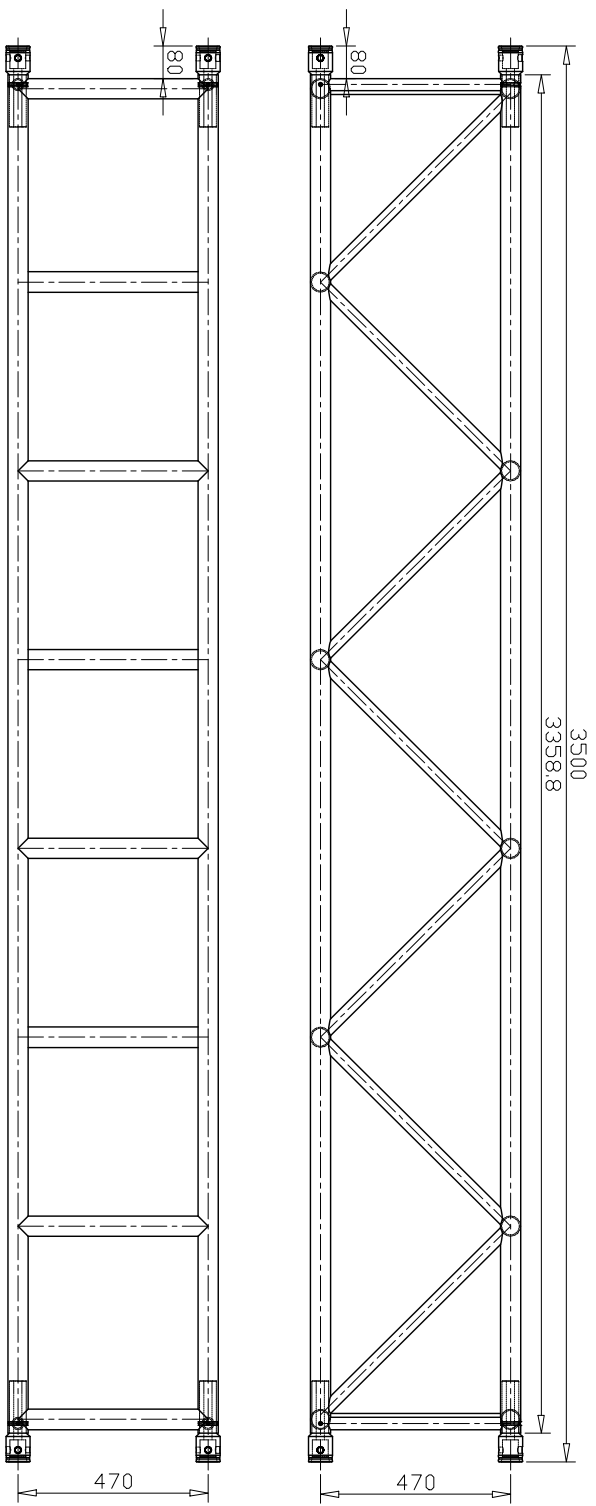


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	#2011	T6		4
	ST-F64-PJ-A-2	Steel	Black	8
	#6082	T6		12
	Ø30*3.0t*617.8mm	#6082	T6	2
	Ø30*3.0t*577mm	#6082	T6	4
	Ø30*3.0t*419.5mm	#6082	T6	4
	Ø50*4.0t*470mm	#6082	T6	9
	Ø50*4.0t*2858.8mm	#6082	T6	4
	ST-F64-PJ-A-1T	#6082	T6	8
ND.	SPEC.	MATERIAL	FIS-	QTY

DRAWN	F54S	DRAW NO	F54S300	2D/3D	edition	2	
Specification	F54S-3.0M	DESIGN	dingkwi	DATE	09.06.29	CHECKED	APPROVED
Weight	**kgs/lbs	UNITS	mm	Proofread			
SURFACE	RAW	SCALE	1:10				



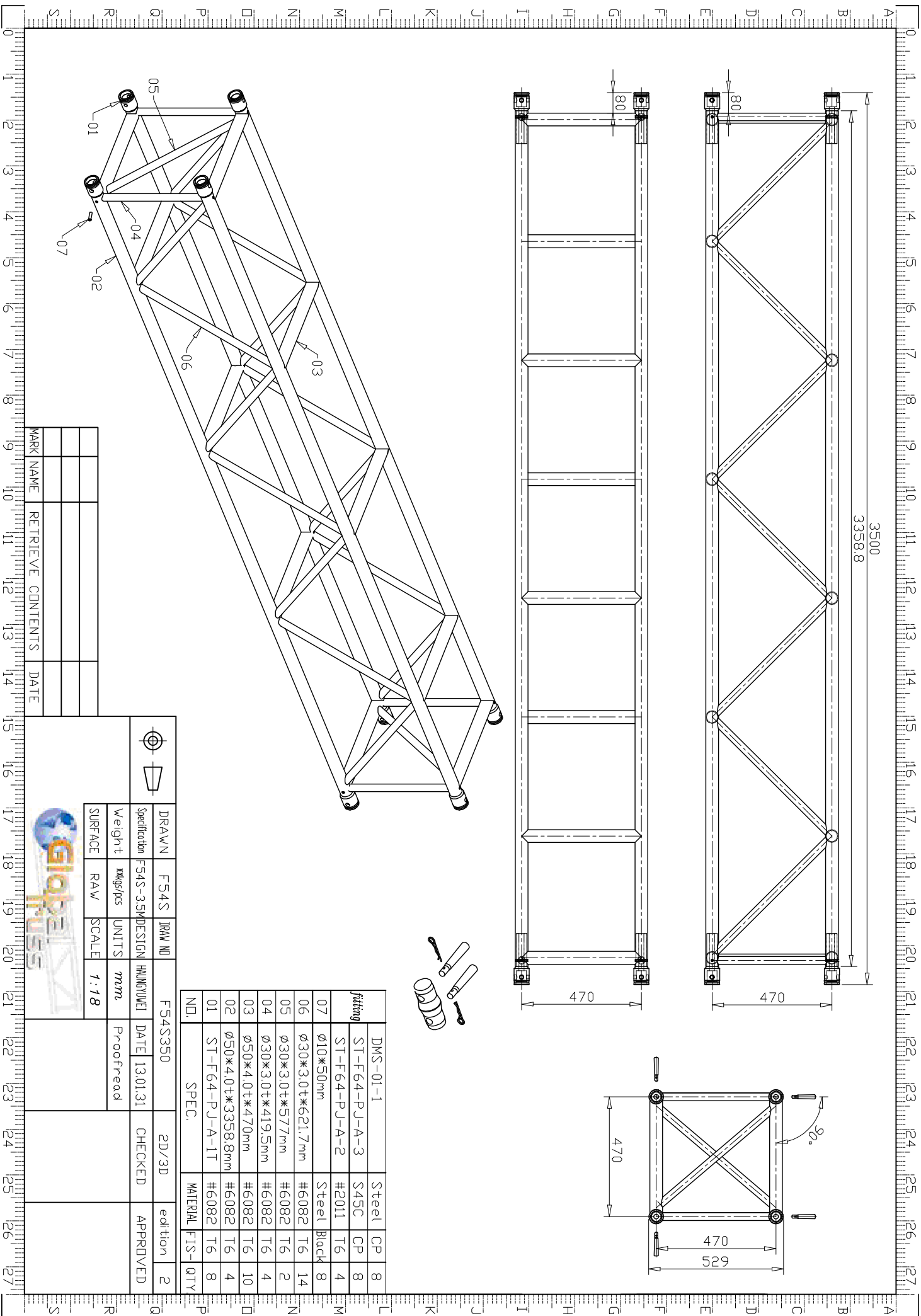
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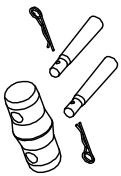
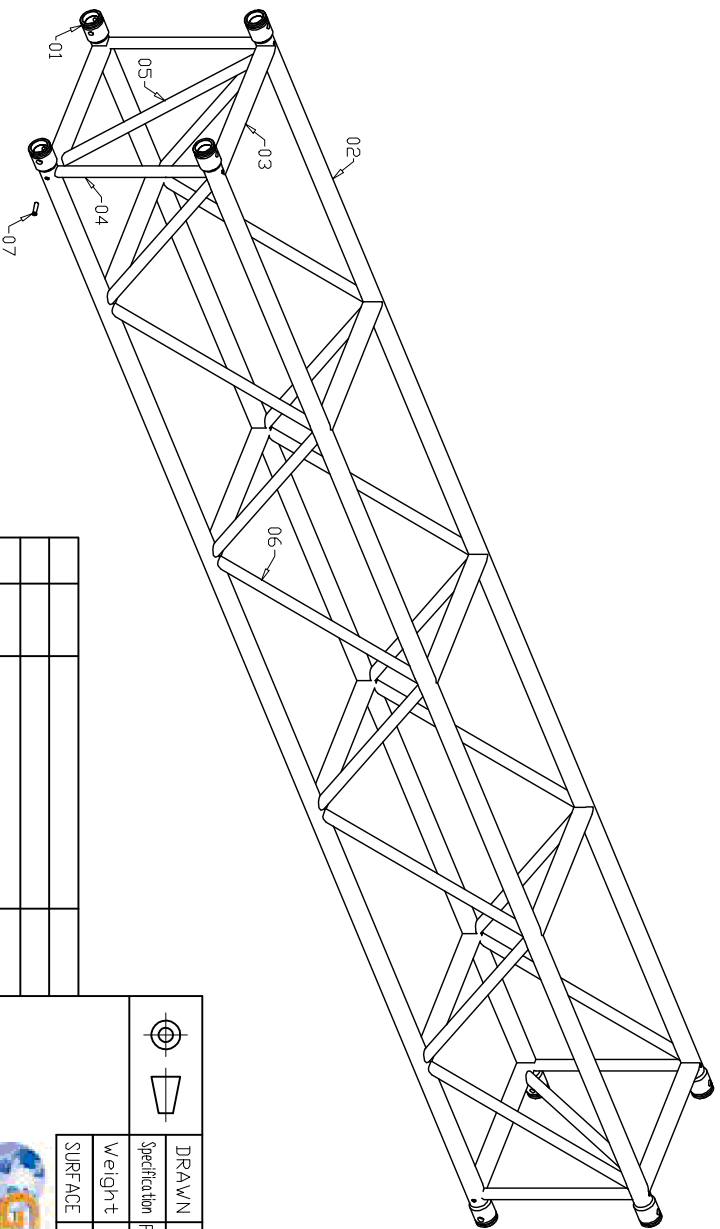
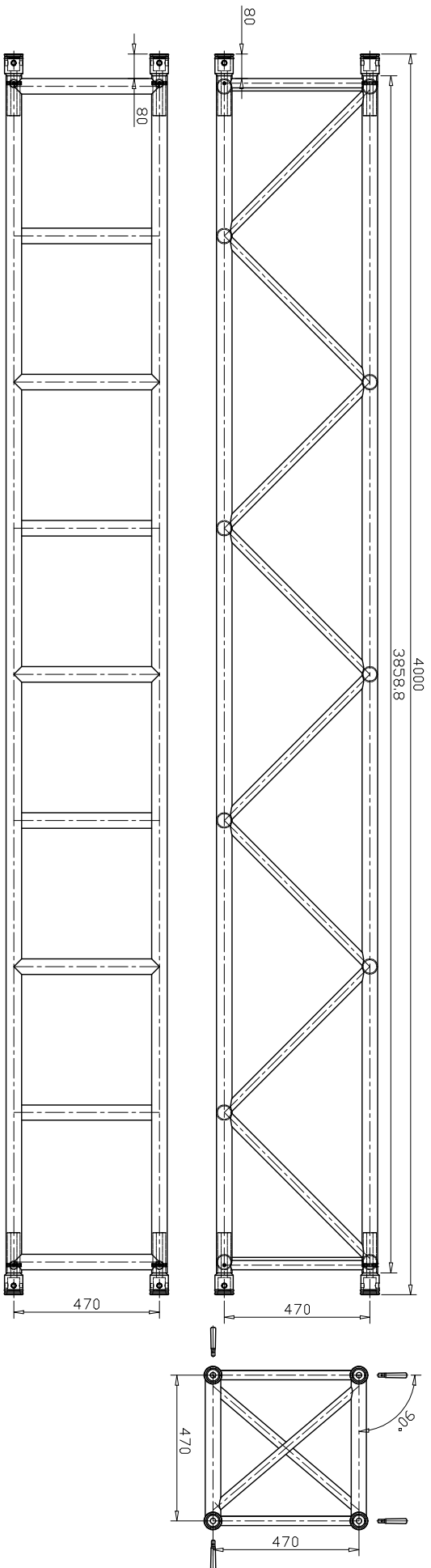


fitting	DMS-01-1	Steel	CP	8
	ST-F64-PJ-A-3	S45C	CP	8
	#2011	T6		4
	ST-F64-PJ-A-2	Steel	Black	8
	Ø10*50mm			
	Ø30*3.0t*621.7mm	#6082	T6	14
	Ø30*3.0t*577mm	#6082	T6	2
	Ø30*3.0t*419.5mm	#6082	T6	4
	Ø50*4.0t*470mm	#6082	T6	10
	Ø50*4.0t*3358.8mm	#6082	T6	4
	ST-F64-PJ-A-1T	#6082	T6	8
ND.	SPEC.	MATERIAL	FIS-	QTY

DRAWN	F54S	DRAW NO	F54S350	2D/3D	edition	2
Specification	F54S-3.5MPDESIGN	HANDLING	DATE	13.01.31	CHECKED	APPROVED
Weight	**kgs/lbs	UNITS	mm	Proofread		
SURFACE	RAW	SCALE	1:18			

MARK NAME	RETRIEVE	CONTENTS	DATE



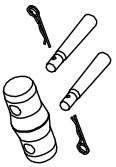
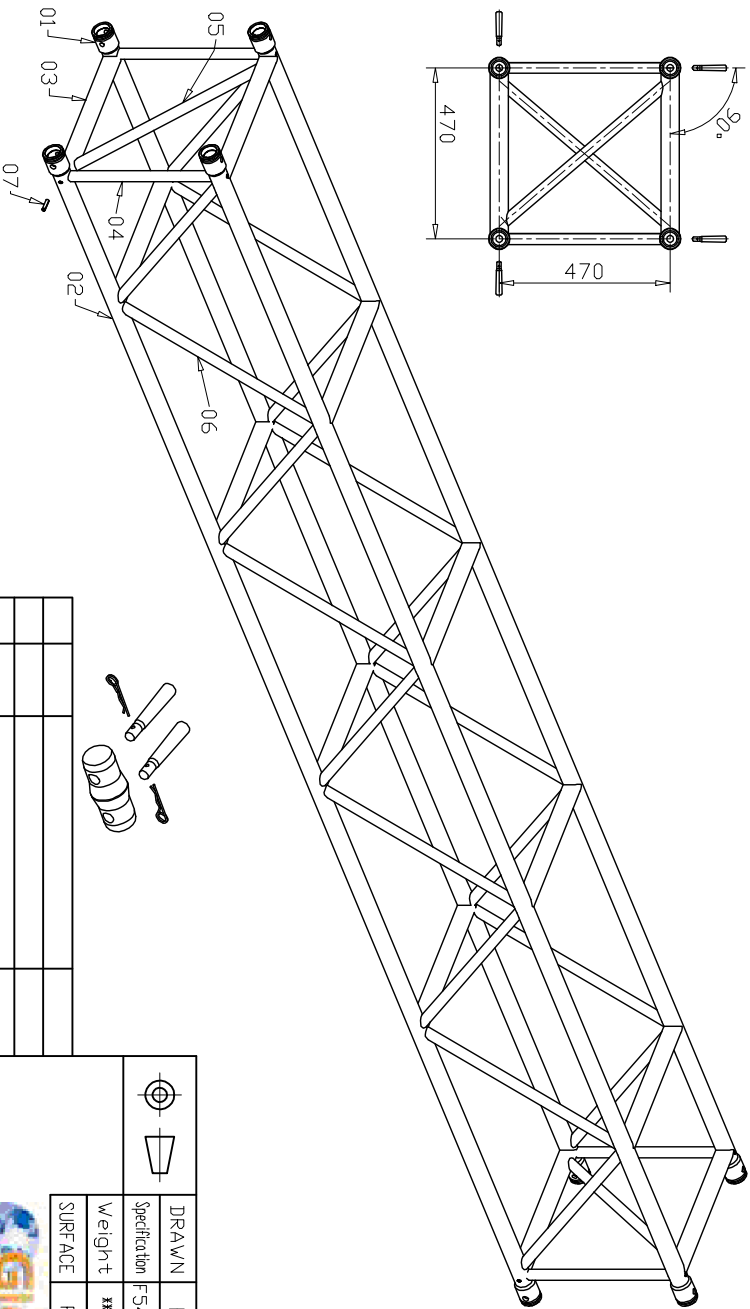
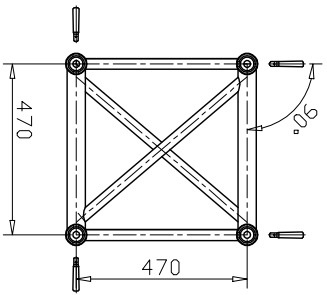
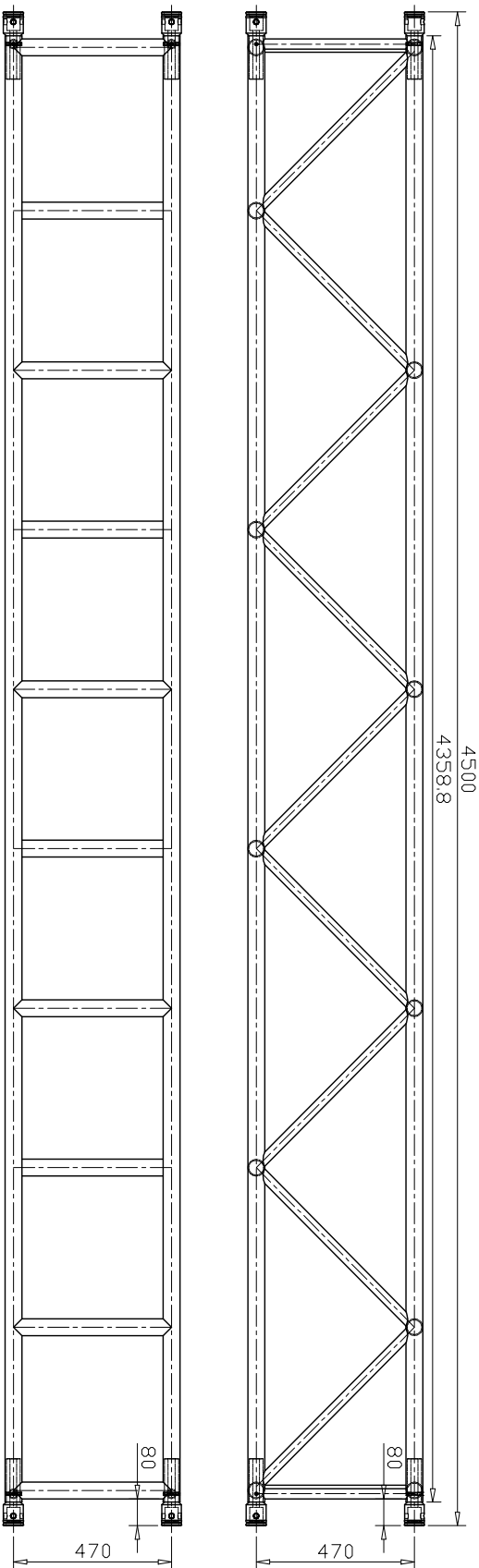


fitting	DMS-01-1	Steel	CP	8
	ST-F64-PJ-A-3	S45C	CP	8
	ST-F64-PJ-A-2	#2011	T6	4
	Ø10*50mm	Steel	Black	8
	Ø30*3.0t*624.6mm	#6082	T6	16
	Ø30*3.0t*577mm	#6082	T6	2
	Ø30*3.0t*419.5mm	#6082	T6	4
	Ø50*4.0t*470mm	#6082	T6	11
	Ø50*4.0t*3858.8mm	#6082	T6	4
	ST-F64-PJ-A-1T	#6082	T6	8
ND.	SPEC.	MATERIAL	FIS-	QTY

DRAWN	F54S	DRAW NO	F54S400	2D/3D	edition	1	
Specification	F54S-4.0M	DESIGN	HUNGWUVEI	DATE	13.01.31	CHECKED	APPROVED
Weight	**kgs/lbs	UNITS	mm	Proofread			
SURFACE	RAW	SCALE	1:19				



MARK NAME	RETRIEVE	CONTENTS	DATE

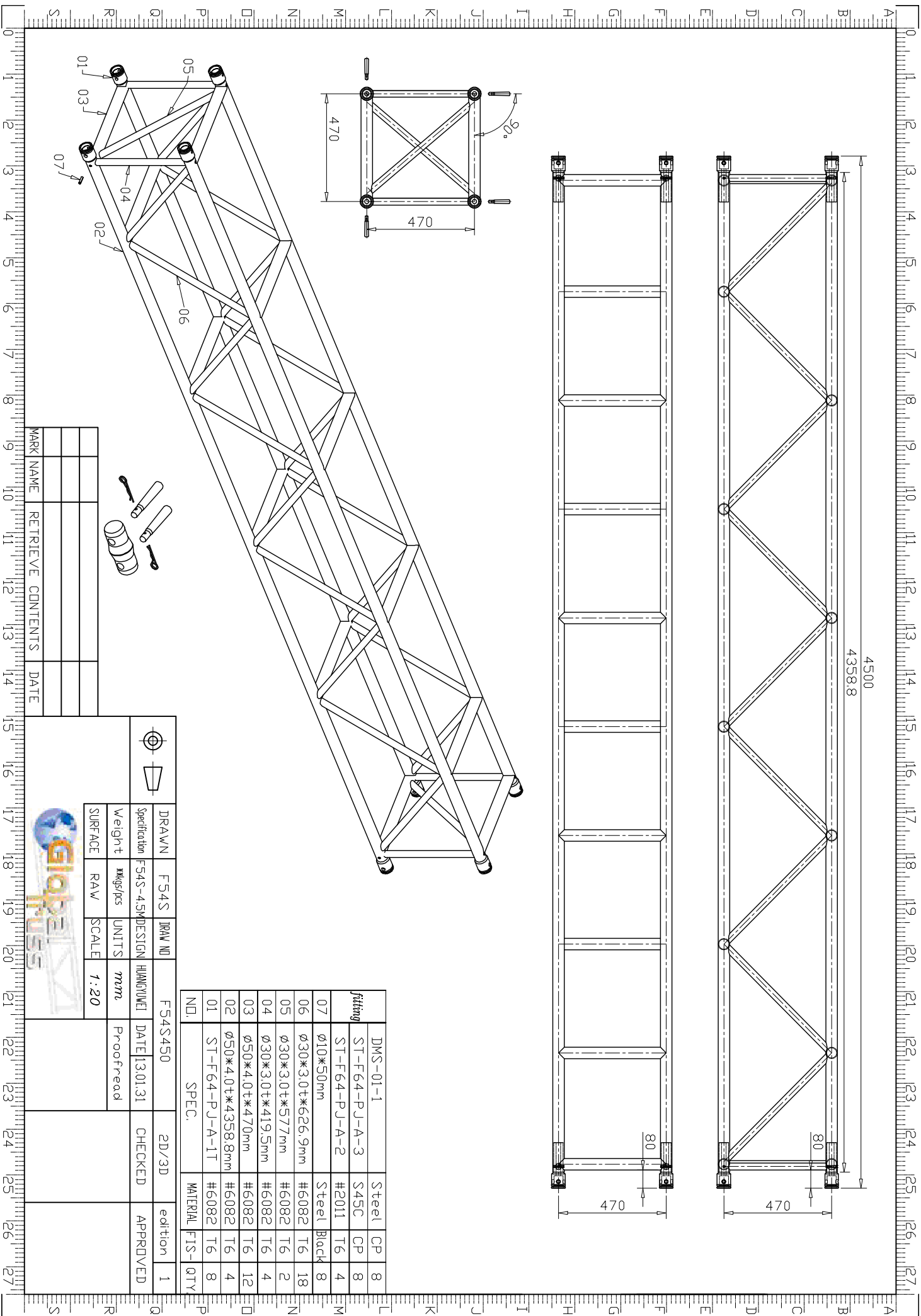


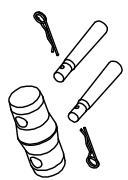
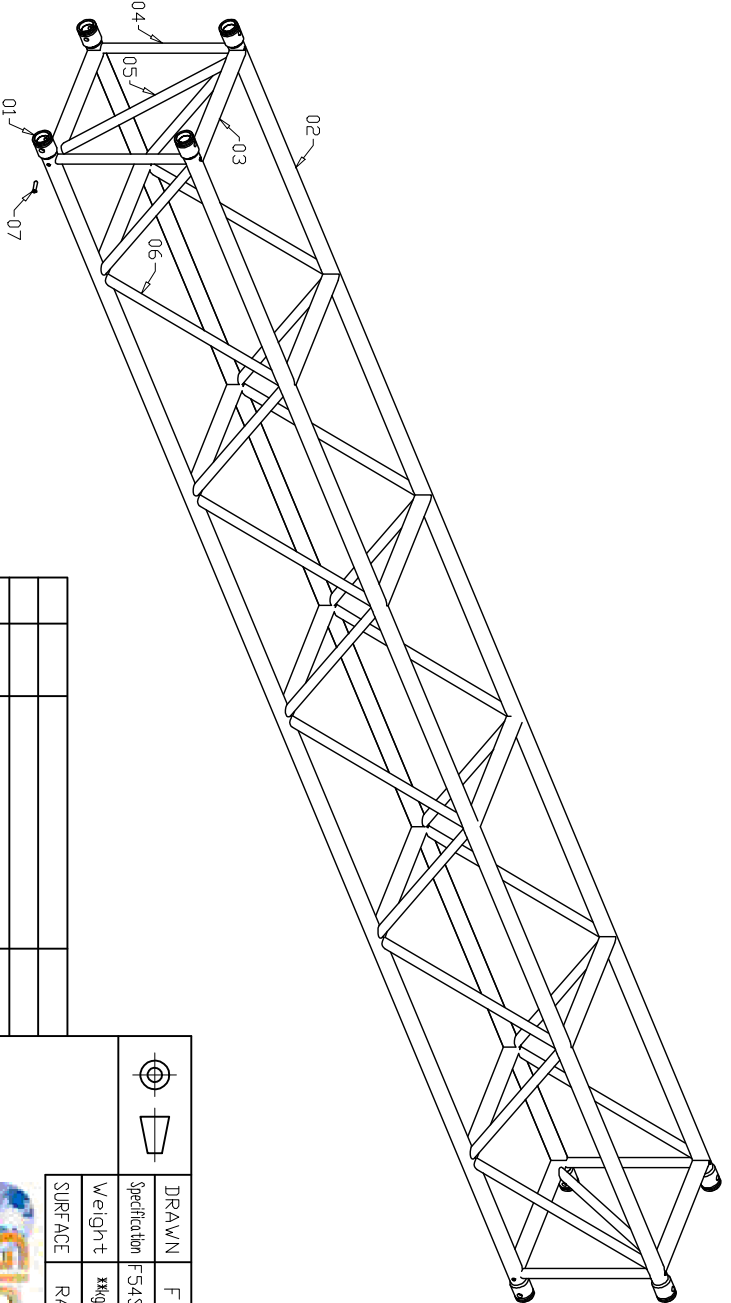
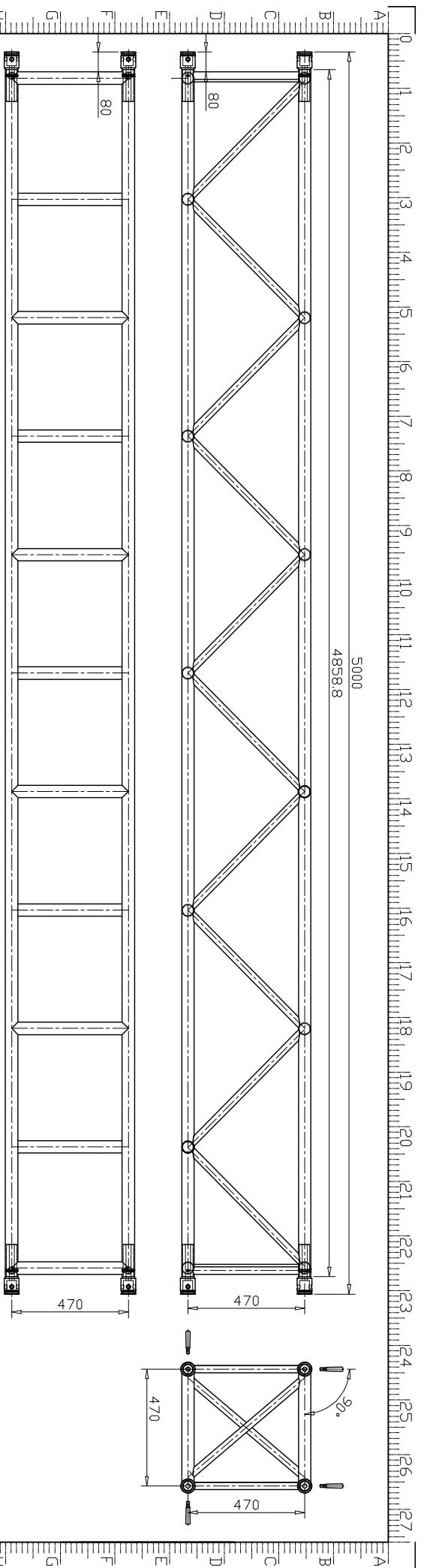
MARK NAME	RETRIEVE CONTENTS	DATE

DRAWN	F54S	DRAW NO	F54S450	2D/3D	edition
Specification	F54S-4.5M	DESIGN		CHECKED	APPROVED
Weight	**kgs/lbs	UNITS	mm	DATE	13.01.31
SURFACE	RAW	SCALE	1:20	Proofread	



fitting	DMS-01-1	Steel	CP	8
01	ST-F64-PJ-A-1T	#6082	T6	8
02	∅50*4.0t*4358.8mm	#6082	T6	4
03	∅50*4.0t*470mm	#6082	T6	12
04	∅30*3.0t*419.5mm	#6082	T6	4
05	∅30*3.0t*577mm	#6082	T6	2
06	∅30*3.0t*626.9mm	#6082	T6	18
07	∅10*50mm	Steel	Black	8





fitting	DMS-01-1	Steel	CP	8
	ST-F64-PJ-A-3	S45C	CP	8
	#2011 T6			4
	Steel Black			8
07	∅10*50mm			
06	∅30*3.0t*628.7mm	#6082	T6	20
05	∅30*3.0t*577mm	#6082	T6	2
04	∅30*3.0t*419.5mm	#6082	T6	4
03	∅50*4.0t*470mm	#6082	T6	13
02	∅50*4.0t*4858.8mm	#6082	T6	4
01	ST-F64-PJ-A-1T	#6082	T6	8
ND.	SPEC.	MATERIAL	FIS-	QTY

DRAWN	F54S	DRAW NO	F54S500	2D/3D	edition	1	
Specification	F54S-5.0M	DESIGN	HUNGUVI	DATE	13.01.31	CHECKED	APPROVED
Weight	**kgs/lbs	UNITS	mm	Proofread			
SURFACE	RAW	SCALE	22				

MARK NAME	RETRIEVE CONTENTS	DATE

